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HUMAN RESOURCES, LOGISTICS AND COST FACTORS IN WEAPON SYSTEM DEVELOPMENT: DEMONSTRATION IN THE FULL SCALE DEVELOPMENT PHASE OF AIRCRAFT SYSTEM ACQUISITION, APPENDIXES B TO B.

By

Gerard F. King

Dynamics Research Corporation 60 Concord Street Wilmington, Massachusetts 01887

Gerard F.// William B./Askren

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LOGISTICS AND TECHNICAL TRAINING DIVISION Wright-Patterson Air Force Base, Ohio 45433

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LABORATORY

BROOKS AIR FORCE BASE, TEXAS 78235

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This final report was submitted by Dynamics Research Corporation, 60 Concord Street, Wilmington, Massachusetts 01887, under Contract F33615-77-C-0016, Project 1959, with the Logistics and Technical Training Division, Air Force Human Resources Laboratory (AFSC), Wright-Patterson Air Force Base, Ohio 45433, Dr. William B. Askren was the Work Unit Scientist for the Laboratory.

This report has been reviewed by the Office of Public Affairs (PA) and is releasable to the National Technical Information Service (NTIS). At NTIS, it will be available to the general public, including foreign nations.

This technical report has been reviewed and is approved for publication.

ROSS L. MORGAN, Technical Director Logistics and Technical Training Division

RONALD W. TERRY, Colonel, USAF Commander SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

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Instructional system development system ownership costing	
20 ASSTRACT (Continue on reverse side if necessary and identify by block number)	
The coordinated human resource technology and the consolidated data development phase of weapon system acquisition. The results of this demovolume consists of Appendixes B through R to that demonstration report demonstration.	onstration are reported in Volume I. This
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TABLE OF CONTENTS

		Pag
INTRODUCTIO	N	3
APPENDIX B	DESIGN OPTION DECISION TREES	5
APPENDIX C	TASK-ORIENTED APPROACH GUIDELINES	17
APPENDIX D	SAMPLE RMCM REPORTS	19
APPENDIX E	PRELIMINARY TASK IDENTIFICATION MATRIX	33
APPENDIX F	USER DESCRIPTION	45
APPENDIX G	TECHNICAL MANUAL/TRAINING TRADE-OFF GROUND RULES	47
APPENDIX H	TASK ANALYSIS WORK SHEETS	49
APPENDIX I	TEST EQUIPMENT AND TOOL USE FORMS	55
APPENDIX J	ANNOTATED TASK IDENTIFICATION MATRICES	59
APPENDIX K	LEVEL-OF-DETAIL GUIDE	63
APPENDIX L	LCOM APPLICATION TO AMST LANDING GEAR WITH FLIGHT LINE SUPPORT EQUIPMENT (SE)	65
APPENDIX M	SAMPLE TRAINING PLAN	99
APPENDIX N	TRAINING PLAN PERFORMANCE OBJECTIVES	111
APPENDIX O	TRAINING PLAN LEARNING ANALYSES	117
APPENDIX P	DESCRIPTION OF MODULARIZED INSTRUCTION	127
APPENDIX Q	PRELIMINARY MEDIA AND ASSESSMENT CHOICES	131
APPENDIX R	SAMPLE JOB GUIDE MANUAL	137

INTRODUCTION

The information presented in Appendices B to R was developed during the demonstration of the coordinated human resource technology (CHRT) and the consolidated data base (CDB) in the full-scale development phase of aircraft acquisition. These data supplement those included in the basic report and provide significant additional detail.

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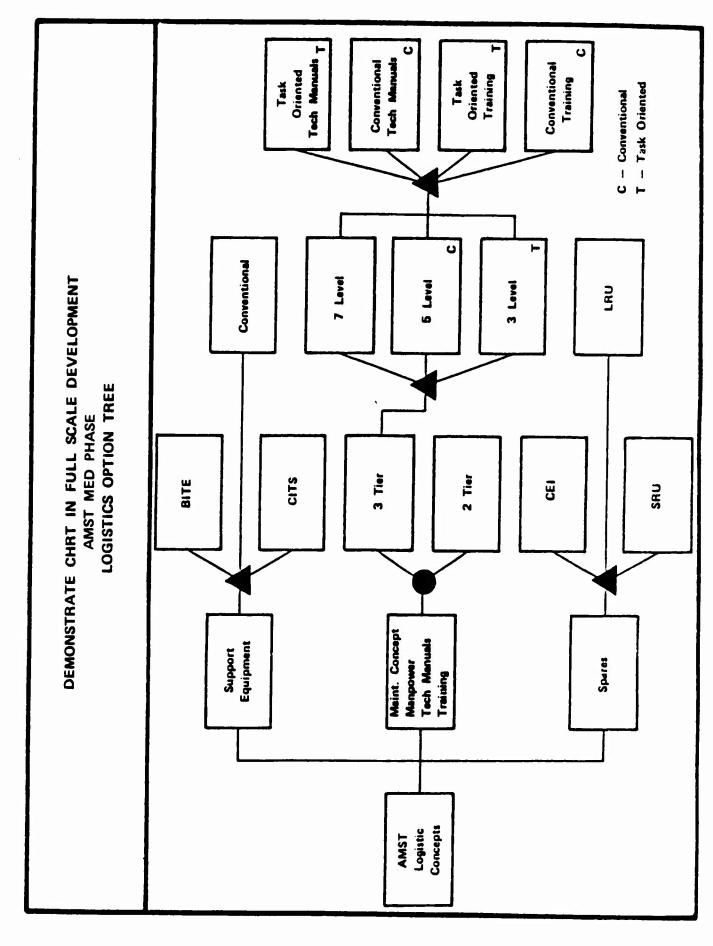
APPENDIX B DESIGN OPTION DECISION TREES

This section lists the full set of Design Option Decision Trees, which were completed during the MED Phase demonstration. Samples are provided.

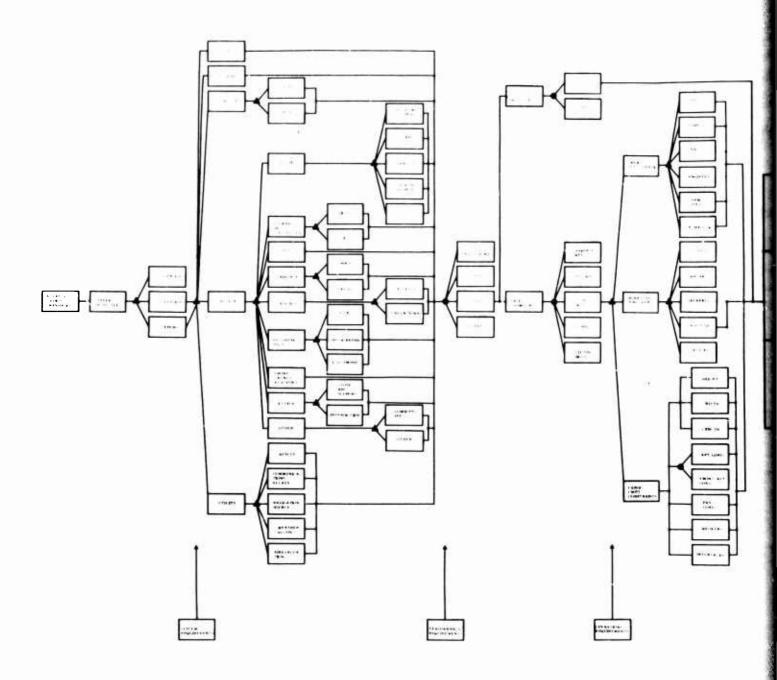
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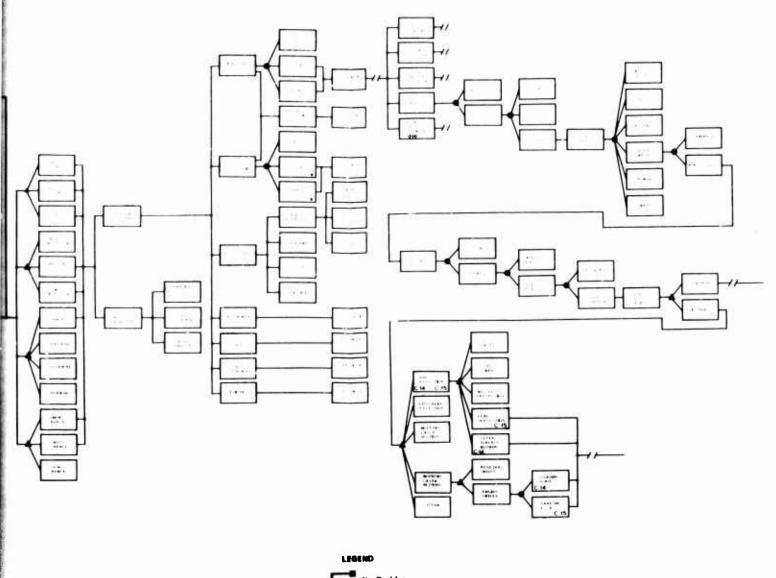
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*1000	AMST System	1	1
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1100	AMST Avionics (ECM)	2	8
1100	AMST Avionics (Radar)	3	8
1100	AMST Avionics (Navigation)	4	8
1100	AMST Avionics (Communications)	5	8
1100	AMST Avionics (Integration)	6	8
*1100	AMST Avionics (Info. Process.)	7	8
1100	AMST Avionics (Inst. & Display)	8	8
*1200	AMST Landing Gear	1	3
*1200	AMST Landing Gear (Main Gear)	2	3
1200	AMST Landing Gear (Nose Gear)	3	3

^{*} Indicates those design option decision trees included in this report.



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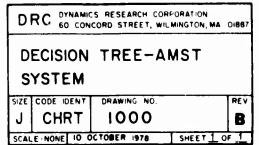




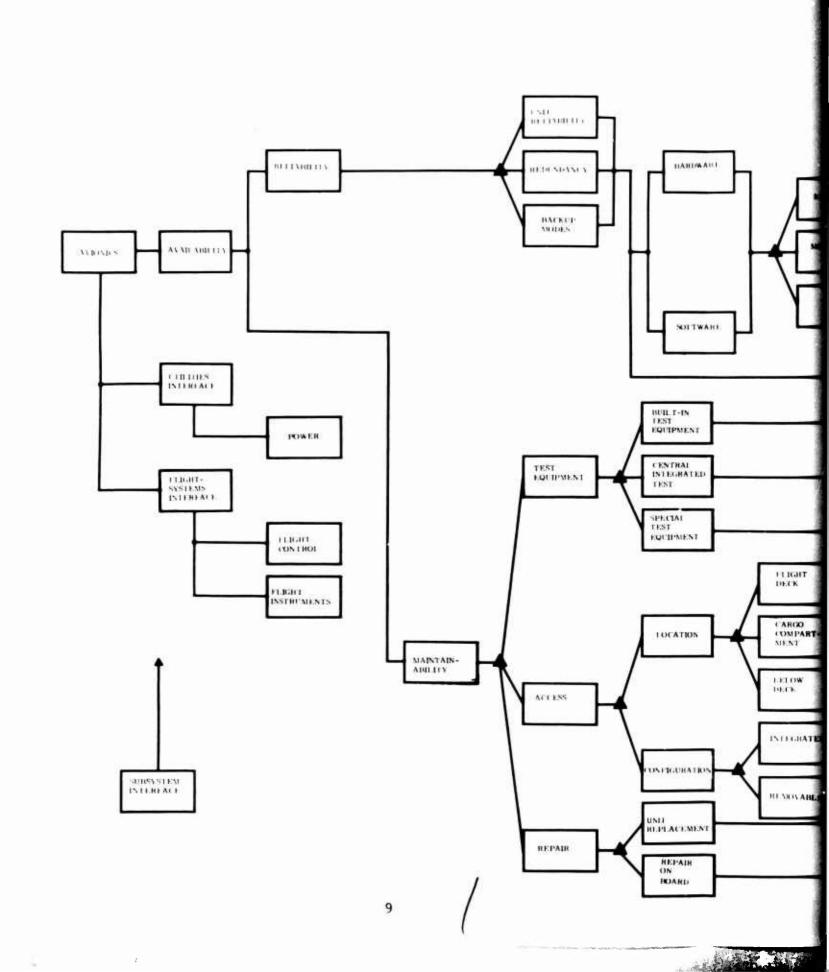


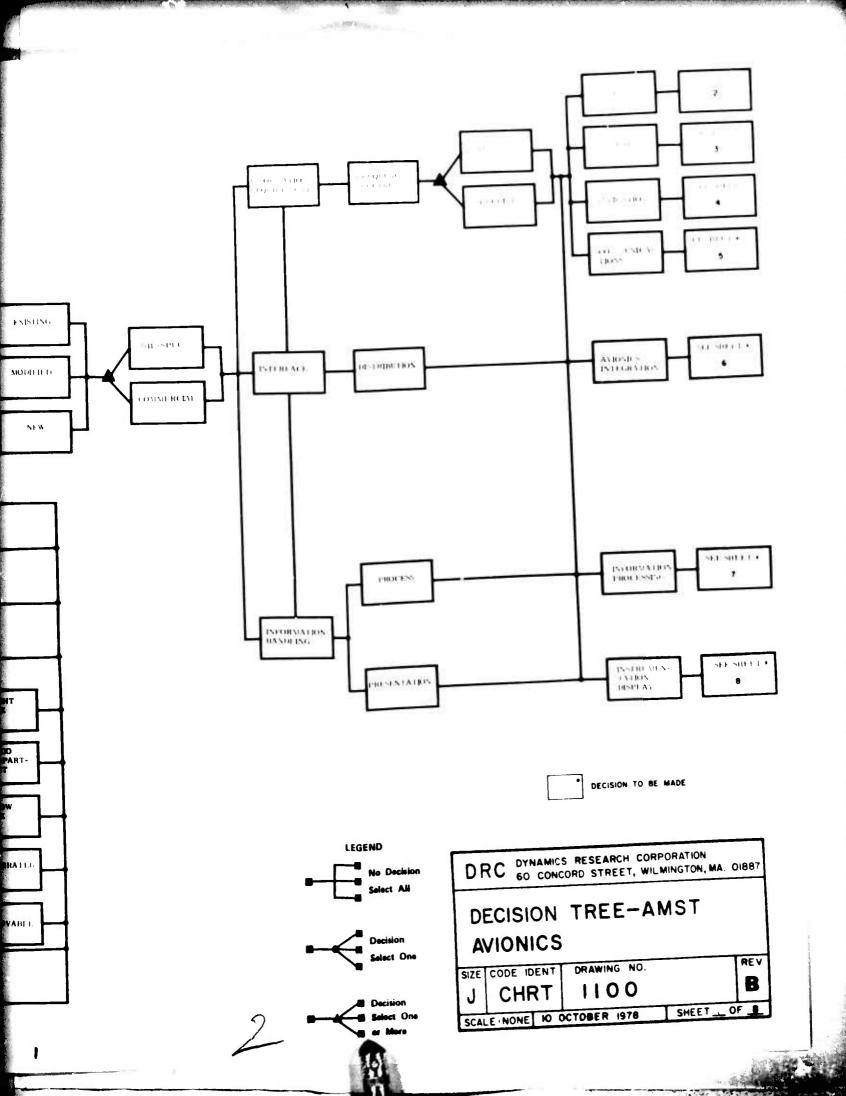


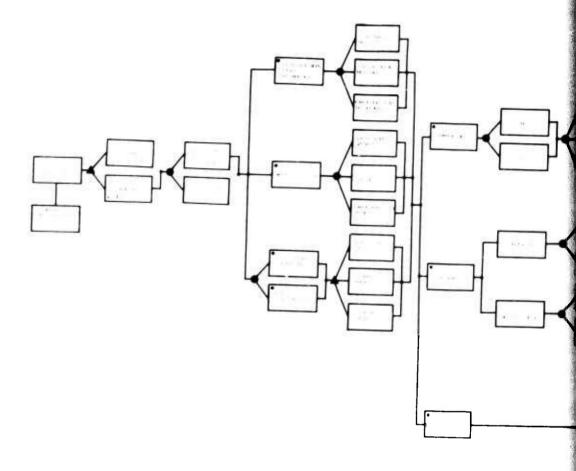


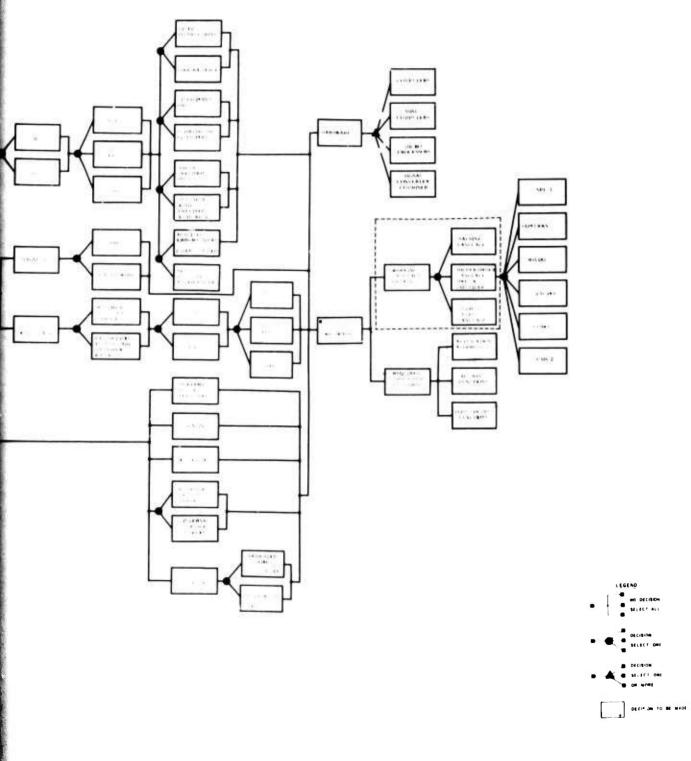




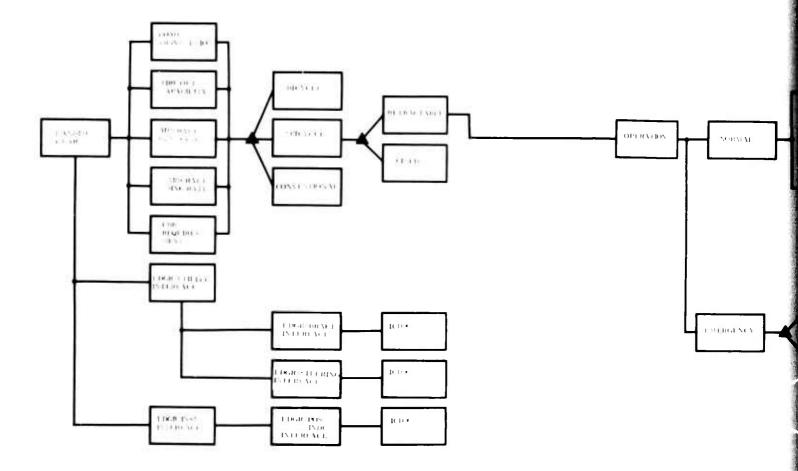


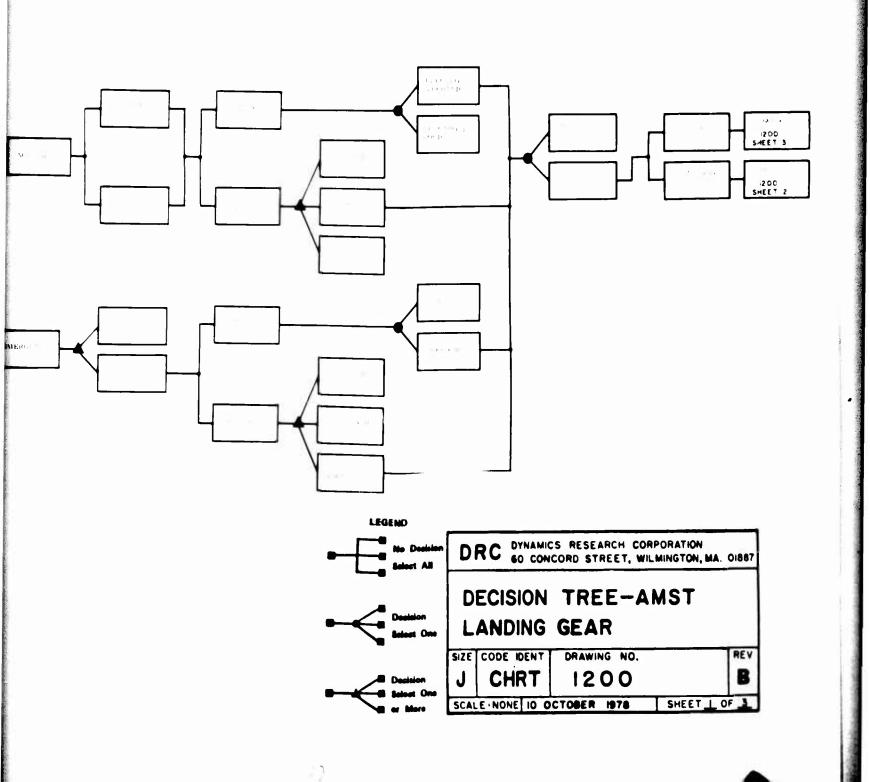


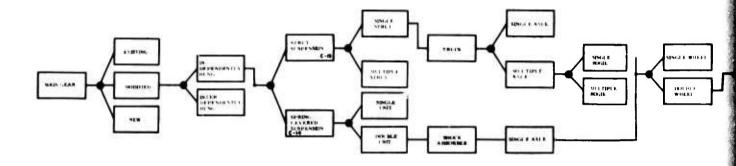


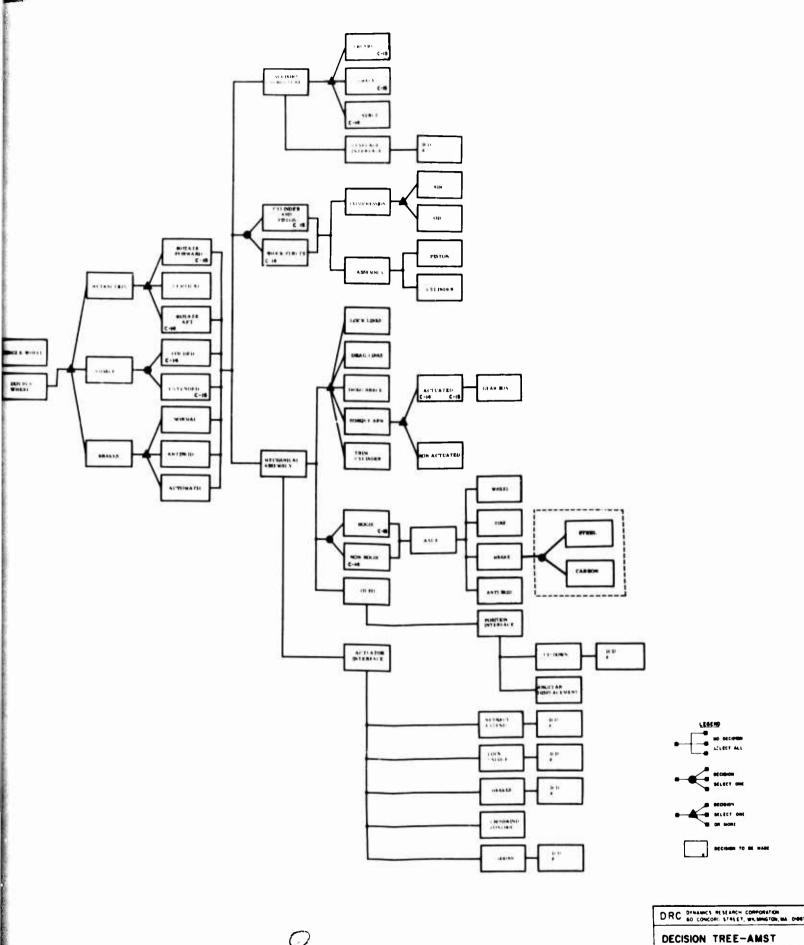


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GUIDELINES APPENDIX C TASK-ORIENTED APPROACH GUIDELINES

Maintenance action networks are prepared to reflect 4 conventional approach to personnel, training, and technical The conventional approach assumes 5-skill-level manning on the flightline with 3-skill-level helpers. These personnel are supported with conventional training and conventional technical manuals. A major system level alternative considered during this demonstration was a taskoriented approach. This assumes primarily 3-level skills on the flightline supported with task-oriented training and task-oriented technical manuals. In order to evaluate this option, the maintenance action networks had to be updated to reflect direct impact of such an approach. This section describes the conservative actions taken and rationale used to accomplish this update.

The action is described and the rationale for each action then follows in parentheses. The rationale has been substantiated by a thorough review of the literature which addresses the implications of the task-oriented approach. The percentages used are representative of conclusions found in the literature.

- Reduce the times for flightline cannot-duplicate, troubleshoot, and maintain an aircraft by 10 percent each. (Proceduralized aids reduce maintenance times.)
- 2. Reduce flightline probability of cannot-duplicate by 50 percent and reduce the number cannot-duplicate actions accordingly. (Produce diagnosis.)
- 3. Increase MFHBMA, as appropriate, based on action (2) above.
- 4. Reduce shop probability of cannot duplicate by 50 percent and reduce the number of flightline remove and replace actions accordingly. (Proceduralized aids reduce false removals.)
- 5. Increase the number of flightline cannot-duplicate

actions by the same number of action (4) to reflect early cannot-duplicate determination. (Proceduralized aids reduce false removals.)

- 6. Retain personnel quantity and AFSC skills, but modify skill levels as follows:
 - a. Assure that one 413X1 is always a 5-level to provide supervision.
 - b. Assure that all shop personnel called to flightline are 5-level (no reduction in shop skills is assumed for this study).
 - c. Set all flightline specialists performing cannot-duplicate, troubleshooting, and remove and replace tasks at the 3-level (proceduralized aids allow jobs to be performed by lesser skills).
 - d. For maintain-on-aircraft acitons and each AFSC involved, require one specialist at 5-level and all others of the same AFSC at the 3-level. (Maintain-on-aircraft is assumed to be a more difficult and complex action. Therefore, skills cannot be reduced.)

APPENDIX D

SAMPLE RMCM REPORTS STANDARD STATION-KEEPING EQUIPMENT

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REPORT NO. 2 -- EXPANDED NON-ERCURBING COSTS (NRC)

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REPORT NO. 3 -- EXPANDED RECURRING COSTS (RC)

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MINTENANCE 29.995,337.0
SUPPORT EQUIPMENT
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MAINTENANCE MANUALS
INVENTORY MANAGEMENT
LCC - TOTALS

REPORT NO. 4 -- COSTS BY SUBSYSTER CONTRIBUTIONS

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2 R C T	194,654.3									
25 R CY 26 R B CY 36 R B B S S 107 - 170 - 1	453,998.6									
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			CPM.	CPM	.
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			2	EARC	
			401A	TOTAL MRC 30.474,774.3	30,474,774.3

REPORT NO. 5 -- COSTS BY LAU CONTRIBUTIONS

RECURRING COST ELEMENTS (PER TEAR)

OUTPUT FILE - AMST STATION KEEPING EQUIPMENT COST DATA - STANDARD

TOTAL	Z BCY	136.029.1	3.834	62.933	3,539.2	24,015,5	0.680	2.071.0	0.050	0.403	1,037.5	0.02	3,317.2			10116.6	44.653	•	ď	•	12,236.0	0.347	. '	; ;	0	•	0.0	000	0.0		;	3,530,010.2	100,000
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400	# BCY	80,718.5	617,674.3	45.826	284,294.3	9,610.7	0.272	1,031.6	620.0	0.178	900	0.003	•	• • • • • • • • • • • • • • • • • • • •			26.648	OTHER RECURRING COSTS CSE		C S V	C.16		CFL	CAC		000	TEST STATION/TEST DRAWER (CSM)		TEST STATION/TEST DRABER (CPT)	CDR OVERHAUL		101AL	
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E ()	M RCV	8.350,0	120,945.7	3.426	212,115.4	2,655.5	0.075	22.9	100.0	0.011	22.9	0.001	107,170,1	3.036		455,098.6	12.861																
# O J	M BC 4	29,585.3	211,687.6	2.997	283,185.3	7.681.3	0.212	510.1	710.0	4.555.8	510.1	0.014	8.829.7	0.2.0	- i	546,125.2	15.471																
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AN253	3,273,665.2	2433 5.589,000.0	\$6.2	8.862.721.
AN254	10.742	1,204,740.0	56.2	1.293.003.0
A N 2 5 5	25.481.7	1,086,750.0	0°00 86.2	1,112,287.
AN256	215,188.6	3,566	0°000 2°°5	1,624,914.
AN257	1,892.9	60.730.0	\$6.2	82,679.
AN253	0.00	0.265	56.2	56.27 56.27
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		1848	SPRTS	3,000,0
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		101	TOTAL NRC	30,474,774.3

REPORT NO. 6 -- RELIABILITY, MAINTAINABILITY, AND AVAILABILITY BY SUBSYSTEM

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	AVAIL	0.89806
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KEEPING EQUIPMENT COST BATA - STANDARD	MITRIKEM	113.506 70.441 3.846 32.051
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OUTPUT	SUBSYS	DAN250 DAN250

MANNOUR COSTS FER YEAR BY AFSC'S AND SUBSYSTERS SUPPORTED

OUTPUT FILE - AMST STATION KEEPING EQUIPHENT COST DATA - STANDARD

AMMUAL BASE FLYING HOURS (ABFH) = 21043.80 HUMBER OF BASES (MB) = 6 PERCENT OF TOTAL LABOR DEVOTED TO BIRECT LABOR (EFF) = 60.00%

707AL COST 240,938,5 8,829,7	615.062.7 107.170.1 722.232.8	28122.73
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1074L LABOR FLIGHTINE (MURF N.M) 1980-624 134.896 2115.520	3980.986	204.987
DIRECT BRN/EN FIGNTLINE (FRNN N.R) 0.05647 0.00385	0.11351	
AFSC SUBSYS (LLR W) 52831	32851	53153
AFSC	32851	53153

REPORT NO. 8A

SPARES REGULATEMENTS -- INVESTMENT

OUTPUT FILE - AMST STATION KEEPING EQUIPMENT COST DATA - STANDARD

NUMBER OF BASES (NB) = 6 ANNUAL PEAK BASE FLYING HOURS (PBFH) = 56116.80 Expected back order (EBO) = 0.10

DEPOT REPAIR CYCLE TIME (BRCT) = 0.17 VRS. BASE REPAIR CYCLE TIME (BRCT) = 0.13 VRS.

107AL COST 70.795.7 1.26.5511.0 545.610.0 4.24.20.0 85.864.8 1.937.046.6 3,000.0 TOTAL ALL BASES...... 1106220279.4 TOTAL 3,000.0 134.280.4 1,076.6 COST OF SRU SPARES 14.280.0 55,701.2 1,108.658.0 4,69.298.6 6,624.4 7,45.0 7,45.0 6,53.1 55.5 OEPOT (LRUBS) 1,705,606.1 COST OF LRU SPARES 82.880.0 24,000.0 16,000.0 3,880.0 3,500.0 27,240.0 (LAUSS) SMOP OTHER SAU (UCSAU) 1000.00 666.67 1293.33 1166.67 1513.33 260.00 7900-00 UNIT COST 6000 24000 18000 3880 3500 4540 260 COC 60180 126.94608 BEPOT SPARES

TOTAL CSP1...... 11.625.288.4

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TOTAL

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SHOP SPARES

SPARES REQUIREMENTS PER TEAR -- REPLACEMENT

OUTPUT FILE - AMST STATION KEEPING EQUIPMENT COST DATA - STANDARD

NUMBER OF BASES (NB) = 6 ANNUAL BASE FLYING HOURS (ABFH) = 21043.80

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4252	0.13690	0.01	0.05	24 000.00	2000.00	26, 289.5		34.259.0
N253	0.07330	0.01	0.05	18000.00	666.67	10,557.1		17.040.9
1254	0.00480	0.01	0.05	3880.00	1293.33	149.0		1.071
4255	0.00000	0.01	0.05	3500.00	1166.67	16.01		
9521	0.00280	0.01	0.05	4540.00	1513.33	101.7		222.8
1257	0.00000	0.01	0.05	260.00	240.00	1.2		1.2
1253		0.0	0.03	0	0		0	0
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TOTAL	0.24510	0.08	0,0	60180.00	7 900 00	38,368.5	15,103.5	\$3.472.0
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REPORT NO. 9 -- SUPPORT EQUIPMENT REQUIREMENTS/COST

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AME	ANNUAL PEAK BASE FLYING MOURS (PRFM) = 56116.80 NUMBER OF BASES (NB) = 6 AVAILABLE ANNUAL OPERATING MOURS (AAOM) = 2080.00	C FLYING	HOURS 6 ING HO	CPSFH	HOURS (PRFH) = \$6116.80 6	0.00					
	TEST S DETAND TIME (TSOEM)		184 E	PATE BASE (A)	COST (0057	800	SE COST/BASE (CPUSE)	INITIAL SE SPARES COST/BASE (CSESM)	COMMECTION COMMECTION HARBEARE (IN)	18VESTRENT COST (CSE1)	
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REPLACEMENT COST (CSE)

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PORT NO. 10 -- COST OF TRAINING

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=	(COJT) (MU) (TRS)	2,512.0 3.94567	1.98742 0.676	2,834.0 0.15364 0.246	TOTAL COST PER BASE
NEEKS AF			30.00 19.7		PER BASE ALL BASES RING CPT (PIUP NG INITIAL CADI
		32851			TOTAL COST TOTAL CPT TOTAL RECURNING

APPENDIX E PRELIMINARY TASK IDENTIFICATION MATRIX

The Preliminary Task Identification Matrix (PTIM) is designed for the purpose of identifying the maintenance task requirements for each item on the equipment list and noting at what level (organizational, intermediate, depot) the task is carried out. The sources of data for developing the matrix are many and varied. Where the matrix is being developed for a new system that is still under development, the analysis has to depend heavily on the systems documentation that is required by the Air Force systems engineering management procedure. When the system for which the PTIM is being developed already exists, the following documents can be utilized:

- Technical Orders and Manuals (TOs and TMs)
- Engineering Reports
- Standard Operating Procedures
- Parts Inventories
- Special Tools and Test Equipment Manuals
- Illustrated Parts Breakdowns
- Other Records

In addition to the above, valuable information can be obtained through interviews with knowledgeable maintenance personnel who are familiar with the equipment. In our case, it is anticipated that TOs and TMs and interviews with knowledgeable and experienced maintenance personnel will be the main source of information for the completion of the PTIM. It should be kept in mind that PTIM is just that, i.e., preliminary or first cut statements of the total set of maintenance tasks that are performed on the system. It is, therefore, subject to refinement and greater detail as the ISD/JGD process develops.

It is obviously critical to identify all hardware items since the omission of an item will result in the omission of tasks and hence they will not be represented in the JGM and

training. The inputs for equipment identification as specified in AFHRL-TR-73-43 (I & II) are as follows:

- Group Assembly Parts List, Numerical Indexes, and the S M & R codes.
- 2. Optimum Repair Level Analysis.

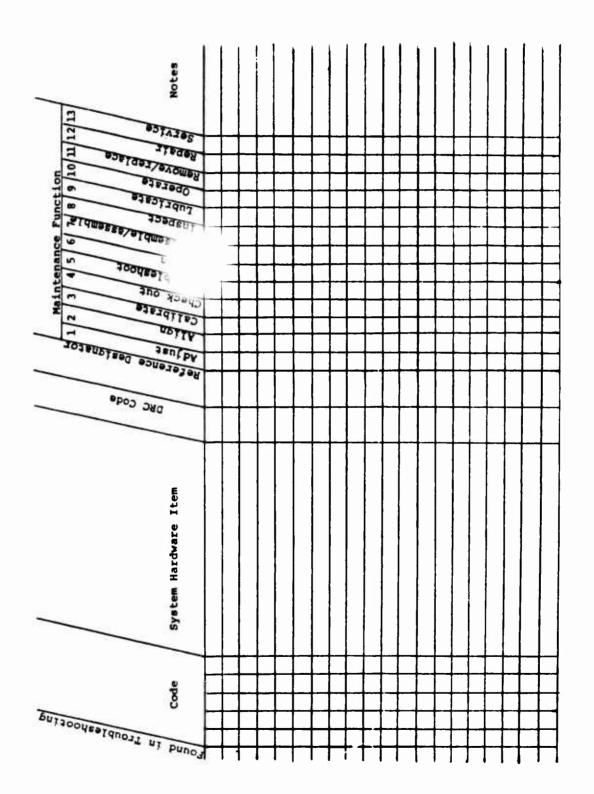
Level of Repair Analysis in compliance with MIL-STD-1390 (Navy).

- 4. Provisionary Lists.
- 5. End Item Maintenance Sheets or Maintenance Sheets system documentation.
- 6. Technical Orders and Manuals.

As a minimum, the preparation of the PTIM requires a Group Assembly Parts List or an Illustrated Parts Breakdown and the associated Source, Maintenance, and Recoverability codes. The format for the PTIM is essentially as shown in Figure E-1. The column and row headings are defined as follows:

Maintenance Function

- 1. adjust to manipulate the equipment in some manner so as to bring it to some specified position or state--usually to bring it from some out-of-tolerance condition to an in-tolerance condition.
- 2. align to bring into precise adjustment or correct relative position by lining up.
- 3. calibrate using special measurements or comparison with a standard, to determine the accuracy, deviation, or variation in a piece of equipment and to correct where necessary.



- 4. check out to perform specific operations to verify operational readiness of the equipment--to test.
- 5. troubleshoot to isolate the source of a malfunction or failure to an item whose parts are replaceable or repairable.
- 6. clean to wash, scrub, or apply solvents to remove dirt, corrosion, or grease.
- 7. disassemble/ to remove and replace the parts of an item for purposes of inspection, cleaning, repair, or replacement.
- 8. inspect to perform a visual, auditory, or tactile examination or check for specific conditions in order to determine the serviceability of an item by comparing its physical and mechanical characteristics with some standard.
- lubricate to put lubrication on specific locations.
- 10. operate to control equipment in order to achieve the intended function.
- 11. remove/ to interchange an unserviceable
 replaced item with a serviceable one.
- 12. repair to restore an item to operable condition by means other than total replacement of a part.
- 13. service to perform operations required periodically such as replenish consumable supplies, to keep an item in proper operating condition.

Whereas in the format recommended in AFHRL-TR-73-43 (I & II) checkout/troubleshoot is listed as one maintenance function,

they probably should be separated, as one can occur without the other especially in the case of scheduled maintenance.

Systems Hardware Item

The equipment items are listed in the row headings. The titles that are being used in CHRT (e.g., brakes/anti-skid) are adequate for initial efforts at the PTIM; however, at some point, these pieces of equipment must be broken down into finer detail. A simple identification such as Main Landing Gear - Mechanical Parts is not sufficient. The greater detail can be obtained from the Group Assembly Parts List or from the IPB.

Code-Reference Designator

The codes showing subordination should be as specified in MIL-M-008910A (AS) and the codes as well as the reference designator should follow the guidelines suggested in AFHRL-TR-73-43 (I & II). However, it is suggested that there also be a code column carrying the DRC code designator (e.g., GLC-173) since this will facilitate the cross-checking of data within the CDB.

Found in Troubleshooting

A check will be placed in this column when, if the hardware item malfunctions, the malfunction is uncovered by the troubleshooting of a superordinate hardware item. The items checked are those that are replaced or repaired as the result of a particular troubleshooting routine.

One further departure from the PTIM format presented in AFHRL-TR-73-43 (I & II) is suggested. The matrix cells are divided diagonally so that in each cell the notation of level of maintenance can be entered and, when the PTIM is annotated, the head, book, joint notation can also be entered. However, this results in an unnecessarily cluttered display, and a separate matrix for both these notations would be better despite the added pieces of paper.

The cell entries for the PTIM are as follows:

- No maintenance task of this type is performed on the hardware item.
- O A maintenance task of this type is performed at the organizational level.
- I A maintenance task of this type is performed at the intermediate level.
- D A maintenance task of this type is performed at the depot level.
- O A maintenance task of this type is performed at both the organizational and intermediate levels.

PTIMS of several detail levels follow. These were derived from a PTIM for the entire landing gear which is not shown. A PTIM for those hardware items removed and order of removal for the subject task is provided in Figure E-2. Finally, a detailed PTIM is provided in Figure E-3.

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Figure E.2. LANDING GEAR DETAILED PTIM (IN ORDER OF REMOVAL - WHEEL, TIRE, AND BRAKE ASSEMBLY)

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FIGUR 6-3 LANDING GEAR DETAILED PRELIMINARY TASK IDENTIFICATION MATRIX (PTIM) (Task

LANDING GEAR DETAILED PRELIMINARY TASK IDENTIFICATION MATRIX (PTIM) (TASK - REMOVE AND REPLACE MAIN LANDING GEAR BRAKE)

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Figure E 3 (continued)

LANDING GEAR DETAILED ANNOTATE) TASK IDENTIFICATION MATRIX (ATIM) (TASK - REMOVE AND REPLACE MAIN LANDING GEAR BRAKE)

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APPENDIX F USER DESCRIPTION

Various types of user descriptions are discussed in AFHRL-Initially, the Air Force would provide the manual developer with technical a preliminary user description, a statement of the type of maintenance technician expected to be working on the system. describes the technician in terms of aptitudes, experience, and job related skills and knowledges. At the same time the Air Force would specify an existing military population most analogous to the expected technical manual users. contractor would then assess this population to determine differences between the statement of characteristics listed in the preliminary user description and those of the analogous population. This assessment would eventually result in a modified user description, a more complete and realistic statement of the technical manual user.

The resources were not available during this study to implement this procedure. Instead a description of the type personnel who might have been assigned to the AMST was prepared and entitled an estimated user description. The technical manual sample was prepared with this population in mind. The sample estimated user description used is provided as follows.

ESTIMATED USER DESCRIPTION

The Estimated User Description (for the 431X2 career field) is that of personnel being trained with the use of proceduralized technical manuals and task-oriented training procedures. The requirements are as follows:

- Aptitude: A minimum of 50 be obtained in Mechanical, through the use of Armed Forces Vocational Aptitude Battery (ASVAB) scoring criteria for USAF personnel.
- 2. Reading Level: A reading level be obtained of more than 60 using the same scoring criteria.
- Aptitude: Average (ASVAB)

- 4. Average Time in Service 18 months
- 5. Prior Indirect Training 3 months BMT
 Prior Direct Training Half of the personnel have
 the required AFSC and 30 days OJT. The other half
 have eight weeks task-oriented training.
- 6. Prior Military Work Experience Half have 2 years experience with the C-130 aircraft. The other half have no military work experience.

APPENDIX G TECHNICAL MANUAL/TRAINING TRADE-OFF GROUND RULES

which direct set of around rules the technical manual/training trade-off is also developed through the integrated task analysis. These ground rules are similar for all systems but must be reviewed and modified for the particular weapon system and user of interest. These ground rules are initially used in developing the ATIM and continuing the tech manual/training trade-off to lower It is important to note that tasks are covered in both manuals and training but the emphasis is different. For example, the need for test equipment and limits to be observed may be presented in a manual, but the use of test equipment is taught. Simple ground rules which are particularly applicable to task-oriented training and proceduralized manuals follow.

Put into Technical Manuals

- Behavioral sequences that are complex and long and which would put a burden on memory
- Behavioral sequences which would require extremely lengthy training/practice periods to produce sufficiently reliable performance
- Tasks that utilize reference information such as tables, graphs, flow charts and schematics, tolerances, etc.
- Tasks that are aided by the presence of illustrations
- Tasks that involve comple
- Tasks that involve complex discriminations or where similarity of cues cause confusion
- Tasks that are performed under stressful conditions that might degrade performance - except time stress
- Infrequently performed tasks.

- Tasks where the probability of error is high and errors are costly.
- Tasks with branching step structures.
- Where low skill level personnel are used.
- Where turnover is high.
- Where procedures change from time to time.

Put into Training:

- Tasks that are not easily described in book form.
- Tasks that are not easily learned on the job (unless they can be put into tech manuals).
- Tasks that need a great deal of practice for acceptable proficiency.
- Tasks where there is little room for error and the errors are costly.
- Tasks which are performed frequently on the job.
- Tasks requiring nigh speed where the rate stimulus inputs are high and response outputs are high.
- Tasks that are performed under stress especially time stress.
- Where environmental constraints interfere with or prohibit use of aids.
- Tasks performed by a large proportion of individuals in a given specialty.

APPENDIX H TASK ANALYSIS WORK SHEETS*

^{*} Task analysis worksheet samples are provided in this appendix. These cover two tasks: remove brake and install brake. These tasks are accomplished after the wheel and tire have been removed.

TASK ANALYSIS WORK SHEET

Tools & Fourions Equipment Training and JPA Implications	B-1 Initial conditions: Aircraft is jacked up with wheel		ck B-3A Tab must be completely screwdriver out of indent or nut hammer won't loosen. May punch require hammer & punch to get tab completely out of the way.	wrench, loosen spanner wrench contingencies. 4 turns with hammer hammer), B-3c	
Task Description	Remove Brake Remove outer spacer (19) B from axle	Request hydraulic specialist to 1) depressurize #3 hydraulic system per T.O. 1C-141A-2-3JG-1 2) disconnect hydraulic line (4) from swivel fitting (3) 3) cap fitting and plug line	Separate tordue fink from brake With screwdriver, bend lock B- tab on star washer (32) out of the indent on nut (33)	With spanner wrench, loosen nut (33) 3 or 4 turns Tap nut (33) with hammer to free bolt (30)	e nut (33), and bolt	Set torque link out of the way by lifting it up and
Task #	B 1.	.:	в)	G 6	θ	(o)

Task #	Task # Task Description	Notes & Cautions	Equipment	Training and JPA Implications
B4.	With assistant, remove brake (20) from axle	B-4 Caution: Brake must be eased off (not allowed to drag) so as to protect bearing surfaces on axle.		B-4 Trg. Practice on brake removal with explanation of importance of protecting bearing surfaces
		Warning: Weight of brake 250 lbs. Warning: Fingers are in some jeopardy		Trg. Proper method of brake removal to avoid injury to back or fingers
5.	Use brake dolly, transport old brake to the service chariot	B-5 Note: Usually new brake would be picked up from chariot at this point in anticipation of brake installation	brake dolly	
.9	Remove inner spacer (20) from axle			

END OF TASK

Task Description	ion	Notes & Cautions	Tools & Equipment	Training and JPA Implications
Install Brake Using solvent, clean and spacers (19, 21)	Install Brake Using solvent, clean axle (22) and spacers (19, 21)	C-1 Inspect spacers for wear or damage	solvent P-D-680	C-1 Trg. Demonstration with examples of adequate and inadequate cleaning
Clean axle and spacers thoroughly with solvent Wipe off any excess solvent	id spacers th solvent excess			
Apply thin film of axle	ilm of grease to	C-2 Note: Spacers are not greased, just axle	grease MIL-G-81322	C-2 Trg. Demonstration required
Install inner axle (make s way on)	Install inner spacer (21) on axle (make sure it is all the way on)	C-3 Flared end goes on first. Frequently requires turning and jiggling to insure that spacer is all the way on		C-3 Trg. Training of technique to get spacer all the way on and demonstration of visual cues to indicate satisfactory execution
Get new brake ar remove any servi wire attachments	Get new brake and completely remove any service tags and wire attachments	C-4 Brake dolly should be used for transport Warning: Possible injury from weight of brake. Any lifting must involve two people.	brake dolly	C-4 Trg. Safety procedures in lifting and handling
Align rotor a of brake	Align rotor and stator discs of brake		screwdriver	C-5 Trg. Must be trained-too difficult to describe in JPA
With assistant, install onto axle	nt, install brake	C-6 Brake must be fully abutted to shoulder on inner space.		C-6 Trg. Visual cues that indicate satisfactory task performance
		Caution: Brake must be eased on so as to avoid damage to bearing surfaces on axle Warnings: Weight of brake, 250 lbs. and hazard to fingers while handling		Trg. Safety precautions in lifting and handling

Task #	Task Description	Notes & Cautions	Tools & Equipment	Training and JPA Implications
C7.	Install torque link (31) Align torque link with brake	C-7a Bolt holes in link and brake must line up		C-7 Trg. Entire task must be thoroughly trained
Q	Install bolt (30)	C-7b Direction of bolt insertion is critical; threaded end must point toward opposite tire. Flat edge of bolt head must line up with alignment bar on brake	hammer	
		housing. (If not lined up, bolt cannot be all the way through and this creates a hazardous condition.)		
		Threaded end of bolt will protrude 1/32 inch from brake housing		JPA Indication of proper protrusion can be depicted by the thickness
		when instanted correctly. (Bolt may have to be tapped with hammer so it will extend the 1,32 inch)		of a line or by the separation between two thin lines.
(c)	install star washer (32) on end of bolt (30)	C-7c Hole in washer conforms to the shape of bolt end so it can be positioned only in a certain position (i.e., washer cannot rotate on bolt end).		
		If washer won't stay in place on end of bolt, a little grease can be put on brake housing so washer can adhere. (But make certain bolt is protruding sufficientiy.)		

1			Tool. 1.	
Task #	Task Description	Notes & Cautions	Equipment	Training & JPA Implications
(P2.)	Install nut (33), tightening it by hand	C-7d Caution: Make sure star washer remains properly positioned on		
(e)	Using spanner wrench, tighten nut (33)	C-7e Tighten to remove all end play May need to use hammer & punch to	spanner wrench (hammer & punch)	
Ć.	Line up a tab on star washer (32) with an indent on rut (33) by tightening nut		spanner wrench (hammer &	
g)	Using screwdriver, bend tab on star washer (32) part way into nut indent		punen) screwdriver	
Ē	Using hammer & punch, seat tab into indent		hammer & punch	
о. в)	Request hydraulic specialist to: Remove cap from fitting (3) and plug from line (4)			
ତି ହି	Connect line (4) to swivel fitting (3) Bleed brake per T.O.			
ন ক	1C-141A-2-3JG-2 Service hydraulic system per T.O. 1C-141A-2-2JG-5			
(o)	Check adjustment of brake and leak check brake per			-
9. a)	Engage parking brake Depress upper portion of rudder pedals (40) full			
(Q	Pull parking brake handle (39) out			
e) 10.	Release rudder pedals (40) Install outer spacer (19)	C-10 Flared end away		

from brake

APPENDIX I TEST EQUPIMENT AND TOOL USE FORMS

The test equipment and tool use forms are initiated early during the integrated task analysis. Initially, they are used to identify the test equipment and tools required and to describe their functions. Finally, the form is used to document the technical manual and training trade-off for test equipment and tools. Those prepared for the torque wrench and one type of spanner wrench are included.

TEST EQUIPMENT AND TOOL USE FORM

İ	9	Dete
PN 722/089-10 Equipment Number		Analyst
Finestons	JPA Training Trade-Off	g Trade-Off
	Information to be included in JPA	Information to be given in Training
1. Calibrated to 960 inchpounds - to torque axle nut 2. Calibrated to 480 inchpounds - to torque axle nut after checking lockring 3. Calibrated to 600 inchpounds (rectangular head) to check lock ring	1. Steps in tightening nut and ring 2. Steps in aligning lock ring holes	1. How to use torque wrench 2. Importance of keeping lock ring in place while tightening nut 3. Alignment of lock ring holes

TEST EQUIPMENT AND TOOL USE FORM

Spanner Wrench

Date	Trade-Off	Information to be given in Training	1. Identification of wrench 2. How wrench is used to loosen & tighten torque link nut 3. How to align lock tab from star washer with slot in the nut 4. How to discriminate when end play between nut and bolt is removed	
	JPA Training Trade-Off	Information to be included in JPA	1. Use spanner wrench to remove & install torque link nut 2. Directive to align lock tab from star washer with a slot in nut	
PN 2559550 Equipment Number	Functions		Used to remove and install brake torque link nut and bolt.	

APPENDIX J ANNOTATED TASK IDENTIFICATION MATRICES

The ATIM file is established after completion of the PTIM, user description, technical manual/training trade-off ground rules and task analysis worksheet files. These files provide the decision data necessary to prepare the ATIM; the allocation of task coverage to training or technical manuals.

The ATIM is retained in the CDB as hard copy in a format similar to the PTIM. The complete ATIM for the task - remove and replace main landing gear brake is shown.

LANDING GEAR DETAILED ANNOTATED TASK IDENTIFICATION MATRIX (ATIM) (TASK - REMOVE AND REPLACE MAIN LANDING GEAR BRAKE)

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	SYSTEM HANDWARE ITEM	Landing Gear System	Inhain Lambing Goor	Statement Part	Lorenton Read Assessment's	1 Cutter Pm	2 Bb-ri	3 Wester	a transfer	Torque Luis Assembly	Sam Wanhar	2 86wt		Bratan/Ann Shall	Stud Detector	Ornica	Rober Dam	Spensor (Outlet)	Spanner (Increase)
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LANDING GEAR DETAILED ANNOTATED TASK IDENTIFICATION MATRIX (ATIM) (TASK - REMOVE AND REPLACE MAIN LANDING GEAR BRAKE)

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		Without and Ting	MLG Wheels	MLG Wheel Amendy	1 Samp Reng	2 Hab Cap	3 Green Returner Reng	d Feit Grame Reng	5 3 Screws (Stat Detection)	6 Cotter Pres (Lembrang)	7 Axie Nut	E Louis Bring	Green Returner Ring	10 Bearing (Outland)	11 Seaponey		13 Fait Grumm Soul	14 Grass Retainer	(parametr) facusers St
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. SEMAICE REPAIR REMOVE/REPLACE MAINTENANCE FUNCTION STARSHO LUBRICATE LOBOSNI SISASSEMBLE/ASSEMBLE LANDING GEAR DETAILED PRELIMINARY TASK IDENTIFICATION MATRIX (PTIM) CLEAN TROUBLESHOOT (TASK - REMOVE AND REPLACE MAIN LANDING GEAR BRAYE) CHECK OUT CALIBRATE NOITY ISDION REFERENCE DESIGNATOR Figure E-3 (concluded) DNC CODE SYSTEM HARDWARE ITEM \$ 5 \$ 2 \$ 2 \$ 2 MLG Tim 1 2 ~ **CODE** • 6 POUND IN THOUBLESHOOTING

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APPENDIX K LEVEL-OF-DETAIL GUIDE

The level-of-detail guide is described in AFHRL-TR-73-43 as a statement of how detailed the information provided in a technical manual needs to be based on the target audience and what is known about the equipment system. It consists of a set of coverage rules stating what needs to be told the technician and how deeply the technical manual should go into each matter. One might normally expect this guide to be written for a specific subsystem and technical skill. During this effort a specific sample task within a subsystem was being addressed. As a result, the level-of-detail guide was prepared at a greater level of detail than is normally appropriate.

Level-of-Detail Guide

1. Disconnect leveler rod
 Simple statement of steps as they occur in their proper
 code Example:

"using pliers, remove the cotter pin"

"Using wrench, remove nut"

"Put nut and washer back on leveler rod for safe keeping"

These steps will be accompanied by a blow-up diagram which (1) shows the location of the rod and (2) details the assembly.

- 2. Remove valve core and deflate tire Simple statement of steps. The appropriate tire pressure which must be reached before proceeding will be specified. The use of the valve core tool and the tire gauge will be taught through training. Cautions will be listed.
- 3. Remove snapring, hub cap, grease retainer ring, felt grease seal, safety wire, skid detector screws and detector, cotter pin from lock ring, and axle nut. Simple statement of steps with suggestions to keep things in order of removal to facilitate installment. Also inspect for wear.

- 4. Inspect wheel for defects
 The specific types of defects likely to be encountered
 will not be described. They will be covered in
 training: The JPA will only direct the technician to
 inspect.
- 5. Remove axle nut and lock ring Simple statement. The use of the spanner wrench will be taught in training.
- 6. All steps up through removal of the tire are listed in a straightforward manner with accompanying illustrations. The washing, lubricating of seals, rings, etc., will be covered in training. However, the material used will be listed in the book.
- 7. Brake removal and installation
 Steps are listed in straightforward manner. The steps dealing with the torque link will be described in somewhat more detail, especially the cues required to align/install/etc., star washer, bolt, and nut. For example, there will be information on such things as how to discern when the bolt is properly installed, when the nut is tight, etc. These steps will also be covered in training. The cautions involving brake removal and installation will be emphasized. Here again cues will be emphasized such as how much of the spacer should be showing when brake is properly installed.
- 8. Tire installation
 The most difficult part of putting the tire back on is
 the alignment of the rotor and stator discs of the
 brake with the wheel. This will be mentioned in the
 book but it will be presented in training. Again,
 mention will be made of the cues that are visible when
 there is proper installation.

APPENDIX L LCOM APPLICATION TO AMST LANDING GEAR WITH FLIGHTLINE SUPPORT EQUPMENT (SE)

INTRODUCTION

This section summarizes the methodology for inclusion of AGE¹ maintenance networks within the framework of AMST landing gear maintenance networks. The effort involved six steps:

- Selection of AGE pertinent to landing gear maintenance.
- 2. Determination and validation of maintenance task data for the AGE.
- Construction of AGE maintenance networks for LCOM simulation.
- 4. Running the simulations.
- 5. The correction of input data resulting from evaluation of simulation model output products and rerunning the simulation.
- 6. The identification of AGE and manpwor requirements necessary to support AMST landing gear maintenance.

METHODOLOGY

The first step in the effort was concerned with identifying the AGE needed to support landing gear maintenance on AMST aircraft. Design decisions during the prototype phase of the study indicated that the landing gear for C-141 aircraft

The acronym AGE for aerospace ground equipment as used in this section is considered synomous with SE.

would be similar to that used for the AMST. As a result, several bases which maintain C-141 aircraft were each asked to submit a list fo the AGE they used, the number of each type of unit assigned to the base, and the monthly demand rate for each of the units. From these lists, different units were initially identified as being pertinent to landing gear maintenance. Two of these units were late ignored in the study; the hydraulic servicing cart was dropped because it was not used for landing gear maintenance, and the hydraulic test stand was dropped because of low usage. Of the seven units selected for the study, the three powered AGE units were: the AM32A-60 Gas Turbine Generator Set, the AF/M27M-1 Hydraulic Jacking Manifold (gasoline powered), and the NF-2 Light Cart. four non-powered AGE units were the MD-3 nitrogen servicing cart, the 35-ton axle jack, the 30-ton tripod jack, and the 70-ton tripod jack.

A trip to Charleston AFB later revealed that an extended version of the 30-ton tripod jack was used there in lieu of the 70-ton tripod jack so this change was made and data was collected accordingly.

Step two dealt with the collection and validation of maintenance task data for the selected AGE. The initial procedure was to extract the data on the AGE from ABD64-A magnetic computer tapes which are a compilation of AFTO Form 349 reports. AFHRL-TR-77-43, "Predicting Powered Support Equipment and Associated Maintenance Manpwoer Requirements" (Reference L-1) was used as an aid in extracting the data. Tapes for a 6-month period (from around September 1977 to April 1978) were requested from each of five bases. the assistance of AFHRL, some data were gathered on the generator sets and light carts. It seems that only registered AGE is logged on these tapes and no data were found on the nitrogen carts. The data compiled from the ABD64-A tape extraction process were useful insofar as maintenance personnel listings and task times concerned.

Following the extraction procedure, scheduled and unscheduled maintenance task flow diagrams were constructed for each of the AGE, both registered and non-registered. Task data such as personnel, resources, and time were entered on the diagrams using available data from the tape extraction and from experienced personnel within the Air Force and DRC. Subsequently, a visit was made to Charleston AFB in order to verify the diagrams and to fill the remaining blanks. Various maintenance shop personnel were interviewed at Charleston including those at dispatching,

gas cart servicing and repair, powered age inspection and servicing, powered age repair, and wheel and tire. The maintenance supervisors and other personnel in these shops were most helpful and informative with the net result that the task flow diagrams for all units were easily completed upon returning. Operational scenario flow diagrams were also completed at that time.

The third step involving construction of AGE maintenance networks for LCOM consisted of two applications. first application, the LCOM simulation was to be run using the demand rate for the AGE based upon total aircraft needs for a given maintenance base. In the application, the AGE was based demand rate upon particular needs of unscheduled landing gear failures for AMST aircraft. In order to achieve this, it was necessary to integrate the AGE with those of the landing gear.

scheduled and unscheduled maintenance operating networks for each of the AGE units constructed from their associated AGE task flow diagrams and format. "Simulating prepared in Extended Form 11 Maintenance Manning for New Weapon Systems," AFHRL-TR-74-97 (Reference L-2) was used as an aid in constructing the networks. The AGE demand rates for this application (Table L-1) were determined from averaging the data received from the five bases. The assigned number of units of each type of age was determined initially by estimating the landing gear maintenance needs of a 52 AMST wing flying about 17 missions per day. This determination was further based on estimation of the downtime probabilities of the AGE and the requirement for periodic maintenance.

The second application of this step uses much the same networks as the first. However, as was noted earlier, they are integrated through their operating scenario into the AMST landing gear unscheduled maintenance networks, since the AGE demand is based upon landing gear failure rates. At present, LCOM is not specifically designed to handle the type of network construction involving SE as used here in that simultaneous landing gear failures, i.e., two failures on one aircraft, can result in multiple demands for the same resource where only one of an AGE type is actually needed. For example, while the AM32A-60 generator can support an entire aircraft, each landing gear failure will call one The net result is that there may appear to be a higher demand on AGE than exists in reality. A similar problem is present in the maintenance personnel area. These may be ways of getting around this difficulty, but for this specific scenario, a method has not yet been perceived,

Age Type	Number Assigned	Average Daily Demand Rate Per Unit	Average Deily Dispetches Per Age Type	90 Day* 90 Day*	Dery.	180 Day 1 Year 2 Year	,		2 Year*
AM32A-60(A) Generator Set	8	1.33	\$	06				80	,
AF/M27M-1 Jacking Manifold	ιo	9.	8			w		~ ~	. ~
NF-2 Light Cart	5	53.	20		8			•	•
MD-3 Nitrogen Cart	^	14.	w		7			~	
35 Ton	ĸ	9.	8		5		m	ı	
30 Ton Tripod Jack (Normal)	å	11.	-		12				
30 Ton Tripod Jack (Extended)	<u>‡</u>	11.	8		8				

Used in pairs for jacking aircraft

Used in groups of four for jacking aircraft

The number of maintanance actions of given types scheduled for all AGE of that class over 180 day scenario.

TABLE L-1 ASSIGNMENT, DEMAND AND SCHEDULED MAINTENANCE

although an unsuccessful attempt was made. For both parts of the step, individual networks for each SE type have been constructed and placed in Extended Form 11 compatible format. Each of these SE networks was developed from maintenance flow diagrams based on the SE unscheduled and periodic maintenance requirements. The SE networks for a fixed demand rate are presented in Figures L-1 - L-6. The networks used in conjunction with the AMST landing gear failure demand are essentially the same, but without the mission call(200000) node.

In order to perform the LCOM simulation runs, the fourth step of the effort, it was necessary to build a scenario around the maintenance networks. The scenario is based upon a wing of 52 AMST aircraft in which one-third of the planes fly each day. Aircraft are preflight inspected between 0400 and 1500, takeoff between 0600 and 1700, and fly 5-hour missions. Although SE is available for use around the clock, most of the SE maintenance personnel shops operate only one shift from 0800 to 1600 (based upon Charleston AFB This is true of AFSCs 423R5 (powered SE scheduling). repair), 431Y2 (gas cart service and repair), 427X4 (welding 423X0 (electrical repair) in this model. remaining AFSCs, 423W5 (powered AGE service and inspect), 431X2 (wheel and tire) and 603X0 (dispatch are on threeshift duty. The simulation scenarios for this model are run for 180-day segments. Each of the scenarios, presented in Figures L-7 and L-8, includes mission or dispatch and periodic maintenance start times, priorities, allowable delay times and frequency of occurrence. An additional important input to the simulation is that of "18" cards for priority specifications. These cards allow for task preempting and expediting, and for overtime allocation in those shops not on three shifts. The initial simulation runs for both scenarios were made with no constraint on manpower; thus, the primary constraints became the task times required to fix the various units and the limiting number of AGE available.

The simulations, based upon landing gear demand, were run with a modified landing gear failure rate. All failures were reduced by a factor of 5; for example, the mean sorties between maintenance action (MSBA) for a 5-hour sortie of the main landing gear was reduced from 145 to 29. The purpose of this reduction was to increase simulation run turnaround times and to reduce computer processing costs by about 50 percent. While the results of this action would naturally reduce the manpower and resource needs for a 130-day period. it was decided that the main objective of the effort, which was to demonstrate the integration of AGE maintenance

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00010 AM32A-60 GENERATOR SET
                 J0102 D
                               00010 31
  J0101 HT0W01
                                             4 C 1D60
                                                           1TUG
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  J0102 Z00000
                 J0103 S
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                               00010 31
  J0103 HKEEP1
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                                            30 25L 1D60
  J0104 HT0W01
                 J0105 D
  J0105 HSERV1
                 J0106 D
                               00010 31
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                                               C 1D60
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                 J0107 D
  J0107 CALLSA
                 J0108 C
  J0108 HREADY
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  J0122 H0160D
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  J0140 HTUG01
                 J0141 D
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                 J0142 S
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                                              4 25L 1423W5
  AAAO4 VAAOOO
                               AA000 21
                                             2 25L 1423W5
 AAA03 RAA000 AAA05 E
AAA05 VAA001 AAA06 D
                 AAAO5 E
                           .20 AA000 21
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                               AA000 21
  AAAO6 SHOP
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                                       AM32A-60 CHASSIS/ENCLOSURE/MOBILITY
 SAAOO1 LAAAOO IAAAO1 E
                           .50 AA000 21
 IAAAO1 WAAAOO
                           .79 AA000 23
                                            34 25L 2423R5
                                            10 25L 1423R5
 IAAAO1 NAAAOO PAAAOO E
                           .21 AA000 23
 IAAAO1 QAAAOO
                      D
                               AA000 21
 PAAAOO PDEPOT
                       D
                               AA000 43
                      Н
                               AAOOO
                                        AM32A-60 ELEC PWR GEN & CONTROL
                           .18 AA000 21
 SAA001 LAABOO IAABO1 E
 IAABO1 WAABOO
                           .66 AA000 23
                       Ε
                                            43 25L 2423R5
                                            23 25L 1423R5 1423X0
20 25L 1423R5
 IAABO1 KAABOO
                       Ε
                           .03 AA000 23
IAABO1 NAABOO PAABOO E
                           .31 AA000 23
 IAABO1 QAABOO
                               AA000 21
                       D
                               AA000 43
 PAABOO PDEPOT
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                                       AM32A-60 GAS TURBINE ENGINE
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 SAAOO1 LAAEOO IAAEO1 E
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 IAAEO1 WAAEOO
                                            25 25L 2423R5
                       Ē
 IAAEO1 NAAEOO PAAEOO E
                           .15 AA000 23
                                           160 25L 1423R5
 PAAEOO WAAEO1
                               AA000 23
                                            30 25L 2423R5
 IAAEO1 QAAEOO
                       D
                               AA000 21
PAAEOO PDEPOT
                               AA000 43
                       D
```

Figure L-1 AM32A-60 MAINTENANCE NETWORKS

```
NF-2 LIGHT CART
                              00010
                     Н
 J0401 HT0W04
                J0402 D
                              00010 31
                                            3 C 1NF2
                                                         1603X0 1TUG
                J0403 S
                              00010 11
 J0402 Z00000
                J0404 D
                              00010 31
                                           50 25L 1NF2
 J0403 HKEEP4
 J0404 HT0W04
                J0405 D
                              00010 31
                              00010 31
                                          4 25L 1NF2
 J0405 HSERV4
                J0406 D
                                                       1423W5
 J0406 DCRMTD
                J0407 D
                              00010 31
                J0408 C
                              00010 31
 J0407 CALLSD
                              00010 31
 J0408 HREADY
                      D
                              00010
                                     NF-2 90 DAY PERIODIC
                     Н
 J0420 HTUG04
                J0421 D
                              00010 31
                                            3
                                                C 1NF2
                                                        1603X0 1TUG
                J0422 S
 J0421 Z00000
                              00010 11
                                          160 25L 1NF2
 J0422 H0490D
                J0423 D
                              00010 31
                                                         1423R5
 J0423 HTUG04
                      D
                              00010 31
                     Н
                              00010 NF-2 2 YEAR WHEEL BEARING
                J0431 D
                              00010 31
 J0430 HTUG04
                J0432 S
 J0431 Z00000
                              00010 11
                                                        1423R5
 J0432 H042YR
                J0433 D
                              00010 31
                                          40 25L 1NF2
 J0433 HTUG04
                      D
                              00010 31
                     Н
                              00010 NF-2 2 YEAR CORROSION CONTROL
 J0440 HTUG04
                J0441 D
                              00010 31
 J0441 Z00000
                J0442 S
                              00010 11
 J0442 HPREP4
                J0443 D
                              00010 31
                                          240 25L 1NF2
                                                         1423R5
 J0443 HTUG04
                      D
                              00010 31
                     Н
                              AC200
                                      NF-2 UNSCHEDULED MAINTENANCE
XCALLSD FAC200
                AC201 F
                            5 AC200 21
                AC202 D
 AC201 AAC200
                              AC200 21
                                            2 C 1423W5
                                            3 25L 2423W5
5 25L 2423W5
 AC202 TAC200
               AC203 D
                              AC200 21
                          .90 AC200 21
               AC204 E
 AC203 MAC200
                              AC200 21
 AC204 VAC200
                                            2 25L 1423W5
                      D
 AC203 RAC200
               AC205 E
                          .10 AC200 21
                                           23 25L 2423R5
 AC205 VAC201 AC206 D
                              AC200 21
                                           5 25L 1423R5
                              AC200 21
 AC206 SHOP
               SAC201 D
                              AC200
                                      NF-2 CHASSIS
                     Н
SAC201 LAC210 IAC210 E
                           .19 AC200 21
                          .17 AC200 23
IAC210 WAC210
                                           85 25L 1423R5
                          .83 AC200 23
                                          15 25L 1423R5
IAC210 NAC210 PAC210 E
                              AC200 21
IAC210 QAC210
                      D
PAC210 PDEPOT
                      D
                              AC200 43
                              AC200
                                    NF-2 ELECTRICAL
                          .22 AC200 21
SAC201 LAC220 IAC220 E
IAC220 WAC220
                      Ε
                          .85 AC200 23
                                           37 25L 1423R5
IAC220 NAC220 PAC220 E
                          .15 AC200 23
                                           52 25L 1423R5
                              AC200 21
IAC220 QAC220
                      D
                      D
                              AC200 43
PAC220 PDEPOT
                     Н
                              AC200
                                     NF-2 ENGINE
                          .59 AC200 21
SAC201 LAC230 IAC230 E
IAC230 WAC230
                          .30 AC200 23
                                          34 25L 1423R5
                 Ε
IAC230 J00000 IAC231 E
                          .70 AC200 21
IAC231 NAC230 PAC230 E
                          .96 AC200 23
                                           34 25L 1423R5
IAC231 NAC231 PAC231 E
                          .04 AC200 23
                                          240 25L 1423R5
                   D
IAC231 QAC230
                              AC200 21
                      D
                              AC200 43
PAC230 PDEPOT
                     D
                              AC200 43
PAC231 PDEPOT
```

Figure L-2 NF-2 MAINTENANCE NETWORKS

```
00010
                                        MD-3 NITROGEN CART
                      Н
  J0501 HT0W05
                 J0502 D
                               00010 31
                                                 C 1MD3
                                                           1603X0 1TUG
  J0502 Z00000
                 J0503 S
                               00010 11
  J0503 HKEEP5
                 J0504 D
                               00010 31
                                            10 25L 1MD3
  J0504 HT0W05
                 J0505 D
                               00010 31
  J0505 HSERV5
                J0506 D
                               00010 31
                                             2 25L 1MD3
                                                           1431Y2
  J0506 CALLPE
                 J0507 C
                               00010 31
 CALLPE HRECHG
                           .33 00010 31
                                                           2431Y2
                                             7 25L 1MD3
  J0507 DCRMTE
                 J0508 D
                               00010 31
  J0508 CALLSE
                 J0509 C
                               00010 31
  J0509 HREADY
                       D
                               00010 31
                                        MD-3 90 DAY PERIODIC
                      Н
                               00010
                J0521 D
  J0520 HTUG05
                               00010 31
                                                 C 1MD3
                                                           1603X0 1TUG
  J0521 Z00000
                J0522 S
                               00010 11
  J0522 H0590D
                 J0523 D
                               00010 31
                                            40 25L 1MD3
                                                           1431Y2
 J0523 HTUG05
                       D
                               00010 31
                                       MD-3 2 YEAR PERIODIC
                               00010
                      Н
                               00010 31
  J0530 HTUG05
                 J0531 D
  J0531 Z00000
                 J0532 S
                               00010 11
  J0532 H052YR
                 J0533 D
                               00010 31
                                            50 25L 1MD3
                                                           1431Y2
  J0533 HTUG05
                       D
                               00010 31
                               AQBOO
                      Н
                                        MD-3 UNSCHEDULED MAINTENANCE
XCALLSE FAQBOO
                AQB01 F
                            15 AQBOO 21
                                             5
                                                  C 1431Y2 1603X0 1TUG
  AQB01 AAQB00
                AQB02 D
                               AQB00 21
                                              3 25L 1431Y2
  AQB02 TAQB00
                 AQB03 D
                               AQB00 21
                 AQBO4 E
  AQB03 MAQB00
                                            20 25L 1431Y2
                           .80 AQBOO 21
  AQB04 VAQB00
                       D
                               AQB00 21
                                             2 25L 1431Y2
                 AQB05 E
  AQBO3 RAQBOO
                           .20 AQBOO 21
                                            15 25L 1431Y2
  AQB05 VAQB01
                AQBO6 D
                               AQB00 21
                                             5 25L 1431Y2
 AQBO6 SHOP
               SAQB01 D
                               AQB00 21
                               AQBOO
                                        MD-3 CHASSIS
                      Н
 SAQBO1 LAQBAO IAQBAO E
                           .30 AQBOO 21
 IAOBAO WAOBAO
                       D
                                            20 25L 1427X4
                               AQB00 23
                               AQBOO
                      Н
                                        MD-3 CONTROLS & GAUGES
 SAQBO1 LAQBCO IAQBCO E
                           .70 AQB00 21
 IAOBCO WAOBCO
                       D
                               AQB00 23
                                           100 25L 1431Y2
 IAQBCO WAQBC1
                       D
                               AQB00 23
                                            40 25L 2431Y2
```

Acceptance of the Control of the Con

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00010
                                       M27M-1 HYDRAULIC JACK MANIFOLD
  J0301 HT0W03
                 J0302 D
                                00010 31
                                             3 C 1M27M 1603X0 1TUG
  J0302 Z00000
                 J0303 S
                                00010 11
                                00010 31
  J0303 HKEEP3
                 J0304 D
                                            30 25L 1M27M
                 J0305 D
  J0304 HT0W03
                                00010 31
  J0305 HSERV3
                 J0306 D
                                00010 31
                                             4 25L 1M27M 1423W5
                                00010 31
  J0306 DCRMTC
                 J0307 D
  J0307 CALLSC
                 J0308 C
                               00010 31
                                00010 3;
  J0308 HREADY
                       D
                                00010
                      Н
                                       M27M-1 180 DAY GAUGE CALIBRATION
  J0320 HTUG03
                J0321 D
                                00010 31
                                                C 1M27M 1603X0 1TUG
  J0321 Z00000
                J0322 S
                               00010 11
  J0322 H03180
                 J0323 D
                               00010 31
                                           160 25L 1M27M 1423R5
                               00010 31
  J0323 HTUG03
                       D
                      Н
                               00010
                                       M27M-1 2 YEAR WHEEL BEARING
                 J0331 D
                                00010 31
  J0330 HTUG03
  J0331 Z00000
                 J0332 S
                                00010 11
  J0332 H032YR
                 J0333 D
                                00010 31
                                            40 25L 1M27M 1423R5
                               00010 31
 J0333 HTUG03
                       D
                               00010
                                      M27M-1 2 YR CORROSION CONTROL
  J0340 HTUG03
                J0341 D
                               00010 31
 J0341 Z00000
                J0342 S
                               00010 11
                               00010 31
 J0342 HPREP3
                 J0343 D
                                           200 25L 1M27M 1423R5
 J0343 HTUG03
                               00010 31
                       D
                               AT900
                                       M27M-1 UNSCHEDULED MAINTENANCE
XCALLSC FAT900
AT901 AAT900
                AT901 F
                             4 AT900 21
                AT902 D
                                                 C 1603X0 1TUG
                               AT900 21
                                             2
                                                                   1423W5
  AT902 TAT900
                AT903 D
                               AT900 21
                                             4 25L 1423W5
 AT903 MAT900
                AT904 E
                           .90 AT900 21
                                             5 25L 1423W5
 AT904 VAT900
                               AT900 21
                                             2 25L 1423W5
 AT903 RAT900
                AT905 E
                           .10 AT900 21
                                            25 25L 1423R5
 AT905 VAT901
AT906 SHOP
                AT906 D
                               AT900 21
                                             5 25L 1423R5
               SAT901 D
                               AT900 21
                               AT900
                                       M27M-1 CHASSIS
                      н
                           .18 AT900 21
 SAT901 LAT910 IAT910 E
                                            20 25L 1423R5
 IAT910 WAT910
                           .50 AT900 23
 IAT910 NAT910 PAT910 E
                                            15 25L 1423R5
                            .50 AT900 23
IAT910 QAT910
                               AT900 21
                       D
 PAT910 PDEPOT
                               AT900 43
                       D
                               AT900 M27M-1 HYDRAULICS
                      Н
 SAT901 LAT920 IAT920 E
                           .49 AT900 21
                                            22 25L 1423R5
20 25L 1423R5
IAT920 WAT920
                           .72 AT900 23
 IAT920 NAT920 PAT920 E
                            .28 AT900 23
 IAT920 QAT920
                       D
                               AT900 21
 PAT920 PDEPOT
                               AT900 43
                       D
                               AT900 M27M-1 ENGINE
                      Н
 SAT901 LAT930 IAT930 E
                           .33 AT900 21
IAT930 WAT930
                           .58 AT900 23
                                            25 25L 1423R5
 IAT930 J00000 IAT931 E
                           .42 AT900 21
 IAT931 NAT930 PAT930 E
                           .94 AT900 23
                                            27 25L 1423R5
IAT931 NAT931 PAT931 E
                            .06 AT900 23
                                           320 25L 1423R5
IAT931 QAT930
                       D
                               AT900 21
PAT930 PDEPOT
                       D
                               AT900 43
PAT931 PDEPOT
                       D
                               AT900 43
```

AST

```
00010 35 TON AXLE JACK
                               00010 31
  JOSO1 HEAVES
                J0802 D
                                            3 C 1AJ35T 1431X2
  J0802 Z00000
                               00010 11
                J0803 S
  JO803 HKEEP8
                J0804 D
                               00010 31
                                           15 25L 1AJ35T
  J0804 HEAVE8
                               00010 31
                J0805 D
                J0806 D
  J0805 HSERV8
                               00010 31
                                           1 C 1AJ35T 1431X2
                J0807 D
  J0806 DCRMTH
                               00010 31
  J0807 CALLSH
                J0808 C
                               00010 31
                               00010 31
  JOSOS HREADY
                      D
                              00010 AJ35T 90 DAY INSP, CORR CNTL & LUBE
                     Н
  J0820 HEEVE8
                J0821 D
                               00010 31
                                           2 C 1AJ35T 1431X2
  J0821 Z00000
                J0822 S
                               00010 11
  J0822 H0890D
                J0823 D
                               00010 31
                                           50 25L 1AJ35T 1431X2
  J0823 HEEVE8
                      D
                               00010 31
                               00010 AJ35T 1 YEAR HYD FLUSH & PARTS INSP
                     Н
  J0830 HEEVE8
                J0831 D
                               00010 31
                               00010 11
  J0831 Z00000
                J0832 S
                J0833 D
                               00010 31
  J0832 H081YR
                                           40 25L 1AJ35T 1431X2
  JO833 HEEVE8
                      D
                               00010 31
                     Н
                               AU100
                                     AJ35T UNSCHEDULED MAINTENANCE
CALLSH YNMAO8
                      E
                           .96 AU100 21
CALLSH YMAOO8
               AU10G E
                           .04 AU100 21
X AU100 FAU100
               AU101 F
                            1 AU100 21
  AU101 AAU100
               AU102 D
                              AU100 21
                                           2 C 1431X2
               AU103 D
  AU102 TAU100
                              AU100 21
                                           2 25L 1431X2
  AU103 MAU100
               AU104 E
                           .75 AU100 21
                                            5 25L 1431X2
  AU104 VAU100
                              AU100 21
                                            2 25L 1431X2
                      D
  AU103 RAU100
               AU105 E
                           .25 AU100 21
                                           20 25L 1431X2
  AU105 VAU101 AU106 D
                              AU100 21
                                           2 25L 1431X2
               SAU100 D
  AU106 SHOP
                              AU100 21
                              AU100 AJ35T (RAM) LIFT ASSEMBLY
                     Н
SAU100 LAU110 IAU110 E
                           .60 AU100 21
                              AU100 23
IAU110 WAU110
                      D
                                          40 25L 1431X2
                     Н
                              AU100
                                      AJ35T RESERVOIR & PUMP ASSY
SAU100 LAU120 IAU120 E
                           .30 AU100 21
IAU120 WAU120
                     D
                              AU100 23
                                          30 25L 1431X2
                              AU100 AJ35T CHASSIS
                     Н
SAU100 LAU130 IAU130 E
                          .10 AU100 21
                                          35 25L 1431X2
IAU130 WAU130 IAU131 D
                          AU100 23
                                           5 25L 1427X4
IAU131 XAU130
                     D
                              AU100 23
```

Figure L-5 35-TON JACK MAINTENANCE NETWORKS

```
00010 30 TON TRIPOD JACK (NORMAL)
  J0601 HEAVE6
                J0602 D
                               00010 31
                                             3
                                               C 2TJ30N 2431X2
  J0602 Z00000
                J0603 S
                               00010 11
  J0603 HKEEP6
                J0604 D
                               00010 31
                                            30 25L 2TJ30N
  J0604 HEAVE6
                J0605 D
                               00010 31
  J0605 HSERV6
                J0606 D
                               00010 31
                                             2 25L 2TJ30N 1431X2
  J0606 DCRMTF
                J0607 D
                               00010 31
  J0607 CALLSF
                J0608 C
                               00010 31
  J0608 HREADY
                       D
                               00010 31
                                        30 TON TRIPOD JACK (EXTENDED)
                               00010
                      Н
  J0701 HEAVET
                J0702 D
                               00010 31
                                             3
                                                 C 4TJ30E 4431X2
                J0703 S
  J0702 Z00000
                               00010 11
  J0703 HKEEP7
                J0704 D
                               00010 31
                                            30 25L 4TJ30E
  J0704 HEAVE7
                J0705 D
                               00010 31
  J0705 HSERV7
                J0706 D
                               00010 31
                                             4 25L 4TJ30E 1431X2
  J0706 DCRMTF
                J0707 D
                               00010 31
  J0707 CALLSG
                               00010 31
                J0708 C
  J0708 HREADY
                       D
                               00010 31
                      Н
                               00010
                                        TJ30N 90 DAY INSP, CORR CNTL & INSP
                                             2 C 1TJ30N 1431X2
  J0620 HEEVE6
                J0621 D
                                00010 31
  J0621 Z00000
                               00010 11
                J0622 S
  J0622 H0690D
                J0623 D
                               00010 31
                                            40 25L 1TJ30N 1431X2
                               00010 31
  J0623 HEEVE6
                       D
                      Н
                               00010
                                        TJ30E 90 DAY INSP, CORR CNTL & INSP
  J0720 HEEVE7
                J0721 D
                               00010 31
                                             2
                                               C 1TJ30E 1431X2
  J0721 Z00000
                J0722 S
                               00010 11
  J0722 H0790D
                J0723 D
                               00010 31
                                            40 25L 1TJ30E 1431X2
                       D
                               00010 31
  J0723 HEEVE7
                      Н
                               AU200
                                       TJ30N&E UNSCHEDULED MAINTENANCE
                               AU200 21
 CALLSG CALLSF
                       C
                           .88 AU200 21
 CALLSF YNMA67
                     - E
 CALLSF YMA067
                           .12 AU200 21
                AU200 E
X AU200 FAU200
                             1 AU200 21
                AU201 F
  AU201 AAU200
                AU202 D
                               AU200 21
                                             2
                                                C 1431X2
                                             2 25L 1431X2
  AU202 TAU200
                AU203 D
                               AU200 21
                                            10 25L 1431X2
  AU203 MAU200
                           .50 AU200 21
                AU204 E
  AU204 VAU200
                               AU200 21
                                             2 25L 1431X2
                       D
                           .50 AU200 21
                                            20 25L 1431X2
  AU203 RAU200
                AU205 E
  AU205 VAU201 AU206 D
                                             2 25L 1431X2
                               AU200 21
  AU206 SHOP
               SAU201 D
                               AU200 21
                      Н
                               AU200
                                        TJ30N&E (RAM) LIFT ASSEMBLY
                           .60 AU200 21
 SAU201 LAU210 IAU210 E
 IAU210 WAU200
                       D
                               AU200 23
                                            40 25L 1431X2
                               AU200
                                       TJ30N&E MANIFOLD & HYDRAULIC LINE
                      Н
 SAU201 LAU220 IAU220 E
                           .30 AU200 21
                       D
                               AU200 23
                                            10 25L 1431X2
 IAU220 WAU220
                      Н
                               AU200
                                       TJ30N&E CHASSIS ASSEMBLY
                           .10 AU200 21
 SAU201 LAU230 IAU230 E
 IAU230 WAU230 IAU231 D
                               AU200 23
                                            35 25L 1431X2
                               AU200 23
                                            10 25L 1431X2
 IAU231 XAU230
                       D
```

Figure L-6 30-TON JACK MAINTENANCE NETWORKS

زان					HEHDER: I	n e c	CIF	·7)					
LIST	180			EHOUS I	AGE DATA	III	ĊL	DHIG	DISPATCHES	۲.	PERIODI	C 1	HSPECT
≥0	1	3	0600	160	DSPD60	Ā	-	0			.5	ં	17.59
زاك	1	ڪ	0630	060	DSP060	1	1	0			L.	3	17.44
نائے	1	ن	0700	Deo	DOF 060	1	1	ij.			.5	3	1
<u> 20</u>	1	3	0730 5000	960 260	DSFD60	-	1	Ů.			.5	3	1599
<u>20</u>	1	3	0800	260	DSF260	1	1	0			.5	0	1599
20 0	1	3	0 830 0900	260 260	DSPD60 DSPD60	i	1	0			.5	3	1999
<u>نان</u> دان	1	3	0930	260	DSPD60	1	1	0			.5 .5	٥	1999
20 20	i	2	1000	260	DSPD60	1	1	Ö			• ='	-0	jiyiyi4 Harana
20	ì	23:23:2	1030	260	73PD60	i	1	Ų.			.5	3	1 संस्थान इ.स.स.स
20	i	3	1100	D60	DSF D60	•	i	Ö			.5	3	1999
20	î	ž	1130	160	DCPD60	i	1	ย				-	1999
20	ī	3	1200	D60	DSPD60	i	î	0			7	ن ق	1999
20	1	ž	1230	260	DSF1060	ī	i	õ				3	1999
26	1	3	1300	D60	DSPD60	i	ī	Ö			.5	3	1999
20	1	ž	1330	D60	DSFD60	Ĭ	1	Ö			.5	3	1999
20	1	1	0300	D60	PE160D	1	1	Ü			30	5	<u> Europa</u>
نانے	1	1	1000	HF2	DSPHE	1	1	0			. 5	5	I terana
20	1	1	1100	HF2	DSPNF2	1	1	Ü			.5	5	1000
20	1	1	1200	HEE	DSPHE	1	1	0			. 5	0.000	1999
فاے	1	1	1300	HFE	DOPHEC	4	1	ķi			.5		्रे नुपृत्
20	1	1	1400	NF2	DSPNES	1	1	Û					1 41414
20	1	1	1500	NFE	DSPNF2	1	1	0			. 5	3	ू 'ब'न'ब
0ك	1	1	1600	HF2	DSFHF2	i	1	Ü				3	1990
20	i	1	1700	HF2	DSPHF2	1	1	Ð.			. 5	3	, जिल्ल
26	1	1	1200	M27M	DM27M	1	i	Û			.5 .5	3	: ખૂખખૂ
الزائ	1	1	1800	M27M	DM274	1	i	Ü			V.5	0	1 પ્રવૃત્
213	Ţ	1	0730	MD3	DSPMD3	1	1	0			. [3	3	* 4.4.4
نائے		1	0930	MD3	DSPMD3	1	1	0			.5	3	1 Para
20	1	1	1130	MD3	DSPMD3	1	:	0				Ů.	1 19 19 19 1 10 10 10 10
20	1	1	1330	MDG	DSFMD3	1	1	0				-	1999
20 20	1	1	1530 0700	MD3 AU35T	DOPMD3 DAU35T	1	i	0				90009000000000	Tightigh Tightigh
20	1	i	1030	AJ351	DAU35T	1	1	Ü			. 5 . 5	0	1999
50	1	i	1230	TUSON	DTUSOH	à	έ	0			.5		5444 1444
20	î	î	1230		DT J30E	4	4	0			.5	3	<u> १</u> ५५५
20	ž	i	1000	D60	PE12VR	1	1	0			155	5	jayyy
20	خ	î	0300	NF2	PE490D	i	i	Ů.			30	Ó	6999
20	5	ī	0300	нга	PE421F	ì	i	Ú			155	5	45999
20	6	ī	0300		PE7900	1	1	Ö			30	5	୍ରସ୍ୟୟ
20	7	ī	0800	MD3	PE5900	ī	ī	Ü				50	. 4.4
20	10	ī	9890		PE8900	1	ī	0			80	5	18999
20	11	1	0800		PE690D	1	1	ō			65	-5	Suga
20	13	1	0300	M27M	PE3180	1	1	0			8.0	15	्र <u>्</u> यम्
£0	13	1	0900	D60	PE1PHT	1	1	0			150	-7	وجوبان
20	15	1	0080		PE81YR	1	1	0			81	5	73999
20	17	1	0880	M27M	PEBEYE	1	1	Ü			150	5.7	146999
20	23	1	0880	M27M	PESPNT	1	1	0			150		146999
50	25	1	0030	NF2	PE4PHT	1	1	6			150	7	48999
20	27	1	0800	мэз	PE52YR	1	1	Ü			150	5	104999

This data may be decoded using AFMSMET Report 78-1 "LCOM II Simulation Software Users Reference Guide" of 1 May 1978.

Figure L-7 SCENARIO FOR FIXED DEMAND

E: 106	ÐНТА	FILE	GENERATE.	D FOF 180.	1057	· ·	1	DENT =	LGHGE	LAI	al HG	GEH	E & HGE	E UB
ETHED	MISS	IOH SO	CEHARIO											
20	1	1	0600 AUS	T HOFLT	1	ı	ľ.	S. OH	. 50	c. 0H	3.08		r jeljeje	
20	Ī	1	Office AMS		• 1	-	Ō	5.014		2.0H			1 - 44	
اباے	1	1	0000 Ans	T HOFET	Ī	1	ij.	5. OH		2. UH			1999	
20	1	1	0900 AHS	T HOFLT	1	1	Ď.	5.0H		2.0H			Test est est	
اراغ	1	1	0930 ANS	T HOELT	1	1	Ü	5.04		2. UH		3	1999	
20	1	1	1000 AMS	T NOFLY	1	1		5.0H		C. 0H		3	1999	
20	1	1	1030 AMS	T HOPLY	1	1	Ũ	5.0H	. 511	2.0H	3.0H	3	1 मध्य	
20	i	1	1100 AHS	T HOFLT	i		Ũ.	5.0H	. 514	6.0H	3. OH		1599	
ارباع	1	1	1100 ANS		1	1	Ľ	5. 0H		2.0H		3	1999	
الات	1	1	1200 AMS		1	1	Ü.	5.0H		L.UH		3	1999	
20	1	1	1230 AMS		1		Ü	5. OH		¿.OH			1 किकान	
اللات	i	1	1300 AMS		- 1		Û	5. մե		2.0H			: ५५५	
වර	1	1	1330 AMS		1	-	ļ i	5. UH		2.0H			1999	
20	1	1	1400 AMS			1		5. OH		2.0H			1999	
20	i	1	1500 AMS		1			5.0m		2.0H			1 ज़ल्ल	
20	1	1	1600 AMS		i		Ü	5.0H		2. GH			રૂ ભૂંખવ	
رياتي	1	1	1700 AMS			1		5.08	. 54	2.08		3	र न त्रान	
20	1	1	0300 006			1		. 1	i,	. 5	3.0	1,	리면면면	
20	Ξ	1	1000 - 060			1		• 1	C	. 5	150	5.	_ = JF919	
26	3	1	1700 BdE			1		5.0H		2.06		3	क्र बालाल	
المات	2	1	0800 DHF.			1		• •	0	. 5	10.0		€',4,4,4	
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رباق	7	1	0800 CMD			1		- 1	Ç	-	4.2	L		
الات	10	1		BOT PERSON			Ü	. 1	Ċ	. 5	6.7			
ابائے	11	1		30H PE6900	_	1			Ę	• 5	0.0	5		
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20	15	1		351 FECINE			O O	. 1	Ç	. 5	č.	5	ি এলসের • ১০.১১১	
20	17	1	0800 0021				Ů.	. 1	Ç	5	15.1		146,499	
20	23	1	0800 10121				Ď.	• 1	(- 5	150	1	(<i>च</i> क्षांत्रांत	
اللت	25	1	0 300 134Fa		1		11 31		1.	. [171		<u> </u>	
20	27	1	0800 XMD:	3 PE52YP	À	1	Į1	- 1	I.	. 5	150		104555	

This data may be decoded using AFMSMET Report 78-1 "LCOM II Simulation Software Users Reference Guide" of 1 May 1978.

Figure L-8 AGE SCENARIO FOR LANDING GEAR FAILURE DEMAND

networks within an aircraft maintenance scenario, was still achieved. The original MSBMAs of the landing gear system were reinstated for the final simulations in order to determine the manpower and resource needs as well as provide for a comparison with Reliability and Maintainability (R&M) model mean value outputs. Results of the initial runs indicated (a) for the constant demand scenario, an overall mission success of 67 percent, and (b) for the landing gear demand scenario, a 100 percent-misssion accomplishment.

The fifth step in the effort involved analyzing the output products of each successive simulation run and making changes to the input data in order to create the desired results. The apparent difference between the mission success of the two scenarios is due to the fact that the demand rates of the two are different. While in one case the success is far below that desired, because of an apparent insufficiency of manpower or resources available, in the other, there may be an overabundance of personnel and resources allocated to landing gear and AGE maintenance. In the following paragraphs, the two scenarios are analyzed separately, although it is pointed out that changes made in one with respect to task times, task probabilities, or failure rates are also entered into the other.

On its initial run, the constant demand scenario simulation called AGE indicated an overall 65-percent mission success. Analysis of the output products indicated that it was only the dispatch missions on the "D60" generators and "NF2" light carts that were not totally successful Further analysis of output products showed that post-sortie times for these two units v greater than 24 hours. Since there are 40 demands per c, on 30 D60 units with a failure expected every two demands, it is not surprising that the average 25-hour post-sorties time would yield the results obtained. Some of chis known time is probably due to the fact that repair personnel, AFSC 423R5, work only one shift, so that any maintenance task not completed by 1600 must wait overnight before work is resumed. The 12 percent unsatisfied demand for these AFSCs lends further credence to this. For the second run, the mission scenario was changed slightly to permit a 1-hour delay (originally 30 minutes) on a dispatch call to see if this reduced the number of mission While the mission success percentage for the D60 aborts. went up only .03 percent, the additional half-hour allowed the NF12 enough extra time for repair in the shop for its mission success went up to 100 percent. Average post-sortie, or downtime for the D-60 increased slightly, while for the NF2, it went down to 17 hours.

The next modification to the AGE data input was to change the D60 failure rate from every two demands to every Mission success rose to 62 percent for D60 dispatches as a result of this change. However, this value was still not satisfactory; so as an experiment, the 1600 to 2400 shift was activated with two 423R5s in the repair The result, was that dispatch success increased This value was still again; this time to 69 percent. unsatisfactorily low, so a further analysis of the input data was made. It is pointed out here that the input analysis of the input data was made. It is pointed out here data collected that the input networks were constructed during a visit to Charleston AFB. task times and rates were based upon esti made by maintenance personnel in the appropriate sops on base. The demand rate was also based upon Charleston's demand. Since to be enough units there to seemed meet maintenance demands, it become necessary to look further into the task times and probabilities for the D60. Several possible areas were considered for modification: (a) the probabilities of flight line maintenance maintenance, (b) the mean flightline maintenance time, and (c) the remove-and-replace task times in the shop. Each of these modifications was made in turn, with increasing the mean probability ratio for flightline/shop success; maintenance was changed from 80/20 to 90/10. Flightline maintenance time was reduced from 24 minutes to 15 minutes and the mean removal task time was reduced from 2.2 hours to 1.1 hours because this time might have been duplicated initially in the shop maintenance times. With all these changes included, the D60 mission success went up to 89 percent as shown in Figures L-18 through L-20. At this point, it was deceided that further manipulations of the maintenance data would serve no useful purpose, and runs for this scenario were concluded.

The scenario based upon AMSI landing gear maintenance demand is called "LGAGE." The first simulation run was made after the third "AGE" run. Results, indicated 100 percent mission accomplishment across the board, as shown in Figures L-21 to L-25. While the 100 percent success may have been affected by the reduced landing gear failure rate, it appeared to be due primarily to the unconstrained number of personnel, the more than adequate supply of AGE resources, and only 17 flights per day. Looking at the Personnel report, Figure L-22, it can be seen that personnel utilization was 4 percent at most. An analysis of manpower demand by hour over the 180-day scenario was conducted using the Manpower Martix output products of the run. For the next run, the number of personnel was limited to from one to four per shift for a

				-								
	OPERATIONS	TOTAL	03000	PE 16 00	PE12YR	PEIPNT	DSPHOS	PE5900	PESZYR	DM27H	PE3180	PE32Y
+	NUMBER OF MISSIONS REQUESTED	10655.00	7208.00	99-99	4.00	7.00	904.88	14.84	2.00	368.80	0	2.00
~	NUMBER ACCOMPLISHED	9843.00	6388.00	91.00	8.00	7.00	966.00	16.00	2.86	360.00	5.01	2.00
*	PEACENT ACCOMPLISHED	92.38	48.72-	100-00	180-80	100.00	188.88	188.88	448488	100	I	•
9	NUPBER OF SORTIES REQUESTED	11015.00	7 200.00	90.00	8.00	7.00	980-68	14. 88	2. BB	360.00	00.5	2.11
-		10203.00	6388.00	90-08	8.00	7.88	968.88	14.00	2.0	360.00		2.08
	PERCENT ACCOMPLISHED	92.63	48.72	100-00	100.00	100.00	100-00	100,00	100.00	10.00	100.00	100
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	8						:					
16	NUMBER DE	:	ċ	2	•	•	-	-	:	•	=	ó
	OPERAT. 10 NS	TOTAL	PE3PHT	DSPNF2	PEASOD	PEAZYR	PEAPAT	DA.435.T	PEAGEN	PERLYE	DT.13.6N	PF6900
-	NUMBER OF MISSIONS REQUESTED	10655.00	2.00	1440.00	38.00	*	4.00	360.20	10.84	3.00		12.00
~	NUM 365 - ACCOMPLISHED	9843-00	2.00	1440.00	30.80		7	360.00	18.08			12.00
*	PERCENT ACCOMPLISHED	92.38	100.00	100.00	100.00	100.00	101.00	100.00	100.00	100.00	100.00	100.00
•	NUMBER OF SORTIES REQUESTED	11015.00	2.08	1860.80	38.08	A.08		366.00	10.00	3.80		12.
~	NUMBER ACCOMPLISHED	10203.00	2.00	1440.00	30.00		1. 1.	364.88	10.01	3.86	166.80	12.0
-	PERCENT ACCOMPLISHED	92.63	100.00	100.00	100.00	100.00	118.88	100,00	100.00	488488	100.00	100-00
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13	90		•	:	:	:	:	i	•	;	:	•
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1	OF RAM	•	:	:	:	:	3	÷	÷	:	-	•
٥	NUMBER OF AIR ABORTS (IDTAL).	• 0	-			4	98	4	4	•	4	4
	OPFRATIONS	TOTAL	01.130£	PE7900	OTHER	1						
-	NUMBER OF HISSIONS REQUESTED	10655400	90.00	22.00								
~	NUMBER ACCOMPLISHED	99. 3.00	90.00	22.00	.0							
m	PERCENT ACCOMPLISHED	92.38	100.00	100.00								
4	VIMBER OF SORTIES REQUESTED	11015.00	360.00	22.00	:							
1	MUMAER ACCOMPLISHED	10203.00	360.00	22.00								
	PERCENT ACCOMPLISHED	92.63	100.00	100.00								
=	NUMBER OF MEATHER CANCELS	0.0	•	•	•							
12	NUMBER OF WEATHER DELAYS	•0	.0	•	•							
-	30	-	•	.0								
2	6	:	•	:	:				1			
4.	OF PAN		0	.0	9							
16	TOTAL STORES OF A POST OF STORES											

AIRCRAFT	TOTAL	XA.3357	xD60	XMD3	XM27M	XNF2	XT.J30E	XTJ36N	OTHER		
TO HOUSE	80.00	6.90	30.00		8.00	12.0	12.00	8.08			
16 NUMBER OF AIRCRAFT-DAYS AVAIL	14400.00	00.006	2400.00	1250.00		10 .00 /2	2160.00	1080.00	٠.		
100	90.5		2.16	.67		. 85					
1.00	12-80	F. 0.2	21.71		1.24	14.20	2.98	2.98			
5.50	.0				•	•	•				
120	-12	.35	4	7	.35	44	115	21.			
20 PCT SERVICE + MAITING	20.16	00.	36.99	10.23	20.41	21.95	.00	•			
	65.38	45.32	38.30	81.59	70.97	62.62	96.76	95.84			
22 AVG. AC POST SORTIE TIME (HRS)	10.75	2.11	10.64	5.47	16.39	15.24	3.88	3.66	•		
		14.	1. 20	7.7	7	55	41	-	•		
24 FLYING HOURS	1021.12	37.33	649.82	91.67	36.93	147.92	38.23	19.22	:		
17 NUMBER OF FCF. TASKS FLOWN		4	-		-			•	•		
IN AVG. AC PRESORTIE TINECHES)	ъ.	.30	•	?	.3	.30	.29	62.	:		
1 3 X X O S & 3 d	TOTAL	A23R5	A23N5	423XB	A27X4	A31X2	43172	68310	OTHER		
27 MANHOURS AVAILABLE (100)	2044-80	316.80	432.00	43.20	28.88	432,00	144.00	648.88	:		
8 PEKCENT UTILIZATION	9.02	13.60	114.05	9	A24	24.38	6119	114.51			1
-	184.34	42.71	47.75	:	.07	110.27	8.92	74.61	•		
30 PCT UNSCHED MAINTENANCE	24.19	36.67	Ph. 77	9	111.00	5.34	26455	64.36	•		
31 PCT SCHED MAINTENANCE	75.81	63.33	55.23	:	:	94.66	75.47	93.66	:		
33 MUMBER OF MEN DEMANDED	42555.00	977.00	16106-00	90	4.00	2945.88	17 30 . 8 82	8795.08	•		
124	93.56	87.82	96.66	:	75.00	100.00	88.96	100.00	:		
2	•		9	4	4	4		4	•		
PCT PROV. BY	00.	.20	:	:	:	•	•	•			
5	78.	82			-	4	-		4		
	7.	11.16	. 02	:	25.10	:	11:00	:	:		
- 1	60	.0.		-	-	4	=	•	•		
40 SIMULATED MM PER FLYING HOUR	18.09	4:10	*. 68	:	•	1.01	.87	7.31			
SHOP REPAIR	TOTAL	AAABA	AABBB	AAESS	ACZIA	ACZZB	AC238	AT938	AT928	AT 938	AGGA
44 MO. OF REPARABLE GENERATIONS	:	:	:	:	:	:	•	:	:	:	•
45 PCT BASE REPAIR	•		•	•		:		•		:	•
			;	:	:	:	:	:	:	<u>:</u>	:
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	:	:	:	•	:	:	•	•		:	:
PCT WHITE SPACE	.0	4	3	•	4		•			-	-
NO. OF	:	:		•	•	3	•	•	•		•
			:	:			:	:	•	:	

Figure L-19 FINAL AGE SIMULATION

				£			PERSON FROM		2		72439
SHOP REPAIR	TOTAL	AOSCO	AU110	AULZB	AU130	AUZ18	AUZEG	AU238	OTHER		
A NO. OF REBARABLE CEMERATIONS		4	1	1	4	-		4	4		
45 PCT BASE REPAIR	:	•	:	:	:	:	:	:	:		
PCT OF PUT	3	9	•		Ē		-	•			
AVERAG	•	:	:	:	:	•	:	•			
	4						ď	-	-		
	: :								•		
NO OF ITEMS IN	-		i.	•			-				
NO. OF	•	•		:	:	:	•	:			
***************************************	TOTAL	AAAOO	AABOO	AAEOA	ACZIA	AC228	AC210	ATOLE	A1928	ATOXO	808
	5100.00	300.00	306.00	308.63	800.00	300.00	366.88	300.00	386.00	300.00	X88.0
1	100.00	180.80	100.00								
30 ca	•	•	:	-			•			:	
NUMBER OF	208.00	108.04	29.88			3.80	7.00	2.44	2.41	2.00	•
PCT	100.00	100.00	100.00	106.00	101.00	100.00	100.00	100.00	111.11	111.11	•
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8				•							•
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×1 d d n t	- TOTAL	AOSC	AULLE	AU120	AU136	AUZ 10	AU228	AU230	OTHER		
rol DOLLAR	5100,00	300.00	308-88	30000	388.88	311.11	211/11	310.11	4		
FILL RATE	100.00	•	÷	:	÷	:	:	:	:		
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UMBE: OF	288.00	ċ	÷	•	=	ċ	:	÷	÷		
PCT OFF-THE-SP	100.00		De-	-	BA	4	4	ě	4		
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41	9	D.	4	•	٩	4	4	4	4		
75.	ė	•	:	•	•	:	÷	•	ė		
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63 40. ITE4S ON BACKORCER (EOP)	•	•	•	•	:	:	.	•	•		
	TOTAL	AJ35L	060	FD.S	HZZH	NF.2	T.J30E	1.39M	TUG	JIHER	
	3600.00	100.00	600.009	140.00	100.00	300.00	240.00	120.00	2000.00	•	
EQUIPMENT	1776.90	216.00	296-00	362.50	216.08	646.00	510.48	259.20	4320.00		
EQUIPMENT	1776.00	216.00	1296-00	302-40	216.11	64.9.00	518.48	259.20	4320-00	•	
71 PCI USED-UNSCHED MAINT	90.	0	-00	9.	•	9	•	0.	111		
72 PCT USED-SCHED MAINT	59.65	4.02	21.71	6.90	7.28	14.29	3.82	3.06	1.62	1.	
PCI	93.29	95 96	78.29	93.10	92.72	85.71	96.18	96.96	98.27	-	
74 NUMBER OF BACKORDER-DAYS	•15	•			:	•	-12	•	:	0.	
75 NUMBER OF UNITS DEMANDED.	62947.00	1479,9025	667.00	3943.00	1467.00	587 4. 00	2586.00	936.002	1795.0	:	
76 PCT AVAILABLE	86.66	100.00		0.00	9:0	100.00	34.66	100.00	~	0	
PCT PROV.	•		•	•	•	.0	0.	0.	•		
78 PCT PROV. BY PRESHPTION	:		•	•	•	•	•	0	°	•	
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					•						

FINAL AGE SIMULATION

Figure L-20

•	PE164.0	Contract to the contract of th	PE12YR A.CO	PE1PNT Z.LD	14.80	FE52YR 2.16	FE3160	PE32YP.	PESPNT 2.84	36.9.1
	161.00	100.66	188.08	100.00		100.30	101.80	180.81	-	Tek-B
NUMBER OF SCRIIES REQUESTED 3335.00	3120.00	30.06		7.00	11:0	2.36	5.40	2.0-2	2.46	34 . 64
	100.00	160.00	100.00	100.30	180.86	130.38	100.30	100.00	10.01	1.6.
MEATHER DELAYS 0.	:		•			:	-	:		
ILERT REPLENTSHHENT	0		-	•						
ATTRITIONS 0.	•	;.	٠.	: .	:	:	:		÷	:
AIR ABORTS (TOTAL)	3.3		:	:	-	:	:	:	:	:
TOTAL S N S I I D N S S S S S S S S S S S S S S S S S S	PELOYE	PELENT	0 6 8 9 9 0	DEALYD	066.000	EE7980	97.00			
TEC	27.4	4.30	10.00	3.00	22.88	12.90				
	4.06	4.60	10.00	3.80	22.88	12.90				
MIMBEY OF SOPTIES BEGINSTER 114.04	166.16	160.00	:		100.00	130.00	. : .			
	30.5	4.30	10.00	3.00	22.88	12.00				
	166.00	100.00	188.38	100.00	100.00	108.00				
MEATHER CANCELS	0		=	:		•				
MEATHER DELAYS 3.	3									
ALERT REPLENISHMENT 3.		•	:	:	÷	:	:			
	4	•	-	4	•	4	3.			
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A MINEFO OF A	AT MIMPER OF ALACOART AUTH. (CCD)	TOTAL	AMST	XA.3357	90x	XMO3	XH27H	XNF2	XT J38E	E ST LX	OTHER	
IN NUMBER OF A	WUMBER OF AIRCRAFT-LAYS AVAIL	23769.50	9366.66	907.66	5460.00	5460.00 1250.00	900.00	2786.06	900.00 2780.06 2160.30 1080.0	1000.00	:	
PCT C	PCT UNSCHED HAINTENANCE	F.	.75			:						
2176	SCHEC HAINTENANCE	1.89	30.45	35	1.56	+26	45	58.	12.	.22		
N 100	POT NGRS	•	؛	•	:	•	•	:	:	;		
	The same of the sa	970		7	1	-	1	-	. 21	-	•	
DCT S	PCT SERVICE + MAITING	1.96	3.19	36.	2,33	.00		1.28		55.		
1 PCT O	CT OFFIRMITCHALLY READY	93.02	45.45	99.66	96.08	89.73		97.75	94.79	49.77		
2 AVG. 4C PCS	22 AVG. 4C PUST SORTIE TIPE (HRS)	5.21	3,32	4.74	47.81	4.53	28.89	37.85	4.46	4.47	-	
3 AVG. NO. OF	IC. OF SORTIES! A/L /DAY	**	.33	-	•	10.	1.	•				
4 FLYING HOUFS	594 594 884	15671.63	15656.19	1.36	10.51	1.60	.90	3.86	2.28	1.20		
2 MUPBER OF EL	NUMBER OF ECF TASKS FLOWN		-		į	9	•		-		•	
AVG. AC PRES	TO AVG. AC PRESCRIIE TIPE(HRS)	1.89	2.04	12.	12.	• 30	82.	. 38	12.	.29		

Figure L-21 INITIAL LGAGE SIMULATION

RUN NUMBER LGAGES	PERFOR	HANCE	٦ ٧	SUNHARY	-		PERIOD F	PERIOD FROM 8. TO 188.0	10 1	0.00	LEVEL 2
FERSCHNEL	TOTAL	423R5	42345	423X8	42334	42334	42353	42354	427X4	431X2	431Y2
27 MANHOLIPS AVATIABLE (198)	20883.83	726.80	2160.60	720.08	720.00	728.30	724.16	723.50	7288	^	726.00
28 PERCENT UTILIZATION	1.00	4.01	.31	•	1.42	56.	2.26	3.56	•	. 30	.12
29 MANHOURS USED (100)	200.54	28.85	64.64	4	11.21	6.69	16.26	25, 63	0	6.57	86
39 PCT UNSCHED MAINTENANCE	52.63	65.9	48.70	•	100.00	166.00 148.00	-	•	• 3	5.98	3.30
21 PCT SCHED HAINTENANCE	67.27	91.81	51.30	-	A.		ď	,	A A	94492	96.73
33 NUMBER OF HEN DEMANCED	22480.88	245.00	245.64 2069.00	•	677.88	677.88 511.80	959.88	959.48 1442.00		1642.00	76.00
34 PCT AVATIABLE (PRIME)	92.72	97.96	185.00	BA	63,37	36.83	62.67	71.64	ď	100.00	97.14
35 FCT AVAILABLE (SUBST.)	•	•	•	:	ۓ	•	å	=	•		
36 PCT PROV. BY EXPEDITE		3	-				4	4	4 1		5
37 PCT PROV. BY PREEMPTION		:	•			•	•	•	9	•	•
38 PCT CEMANDS NOT SATIS.	7.28	2.04	B		36.63	65.17	37, 33	28.36	d		2.26
39 OVERTINE MANHOURS USED (1888)	74.	*0.	0	•	20.	. 85	684	. 10	•		ت
AN STRULATED MM PER FLYING HOUR	1.33	410	484	4	187		41.	416	T T	44.4	1.0
								,			
		-		40.44	4 4 4 4 4		****	1000			

PESSONEL	TOTAL	4313R	4313W	43131	4315R	4315H	43151	53135		53154	53155
27 NANHOUPS AVATIABLE (108)	28880.00	726.88	220.68	- 1		729.88	729.88 2166.88	720.84	720 .04	723.68	720.06
26 PERCENT UTILIZATION	1.00	.01	.17		.7.	.23	3.29			.75	.75
29 MANHOLRS LISED (1891)	28A 5 54	416	1.23	7.27	54.31	1.63	74.99	44.2	7.5	•	5.42
30 PCT UNSCHED MAINTENANCE	52.63	166.00	100.00	100.00	100.00 100.00	100.00	1	100.00	100.00	180.00	101.00
11 PCT SCHEC MAINTENANCE	67.37			-	•			-			•
33 NUMBER OF MEN DEMANDED	22465.00	2.08	93.00	448.00	386.00	135.001	0541.00	2.1	19.00	19.86 269.68	271.48
34 PCT AVATIABLE (FRIME)	92.72	114.84	94.62	75.12	62. AA	45.56	45.56 100.00	182.84	104.00	96.65	96.68
35 PCT AVAILABLE (SUBST.)	9.	9	-	•	•	•	•	•		:	•
TE DEL BEDY, BY EXPERITE	0.0	0	D.A.	•			4	2	4	•	-
37 PCT PROV. BY FREENPTION		•	0.	•			•	•	•	:	:
14 PCT CEMANOS NOT SATIS.	7,28	3	5.34	23.44	37.56	41.44	1	1	7.7	30.35	34.32
39 OVERTIVE PANHOURS USED (100)	. 41	٠	6	•0•	.83	•	•	•		•	• •
- 40 SIMULATED NY PER FLYING HOUR	1.33	49	181	116	18.3	184	24.	40.	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	E 4.	. f. 3

A		30.35 30.32	., .,	E 4. 684															
	•	7	•	111															
	:	4	•	244															
1	•	4444	• •	184									!						
	•	37.5	•	9															
	•	34 234		1		6.0	E J												
a de	•	£ . 5.	C .	10	•	STATE OF US	•	.42 0	9.83	6.62 6.	Q3.9A	2699.fB C.	100.40 0.	£.	R. C.	3	4	0.0	410
		7.28	14.	10.33		Tofai	20.0883	1.03	209.54	52.63	67637	22489.60	92.72		2.	1.	7.28	14.	M. C.
TITTER BY STEEL THE	PCT PROV. BY FREENPTION	PCT CEMANOS NOT SATIS.	39 OVERTIVE PANHOURS USED (100)	48 SIPULATED NY PER FLYING HOUR			27 MANHOLDS AVATIABLE (194)	28 PEPCENT UTILIZATION	25 MANHOLDS USED (100)	FC: UNSCHED HAINTENANCE	PCT SCHER MATHTFRANCE	33 NUMBER OF MEN CEMANCED	PCT AVAILABLE (FRINE)	PCT AVAILABLE (SUEST.)	PCT PEOV. BY EXPECTTE	PCT PROV. BY FREEHPTION	PCT DEMANGS NCT SATTS.	39 OVERTIPE PARHOURS USED (160)	A SIMULATED AF PER FLYING HOUR

1												
A. MO.	SHOPREFAIR	TOTAL	OTHERS	AAAGG	AABGG	AAEOC	AC218	AC228	AC238	A1 91 C	AT 920	A: 930
54		100.00	5	189.85	100.80	١	1		100.00	100.001		\$ 3
46				9		d		3	0	3		
47 AVE	AVERAGE BASE REPAIR CYCLE	10.88	0	60.73	•	•			6	69.65	-	
-	PCT ACTIVE REGAIN	10.03		164.05	ď	9	9	3		160.6)		
Ú.	PCT WEITE SPACE	00.	<u>.</u>		.0	•	•	0		£7		3
DN 05	DE ITEMS IN REPAIR CEDEL	57.00	4	Ç	1.00	-	1.11		1.04	3		
£1 NO.	. OF ITEMS BACKLOGGED (EGP)	1.00	0.		•	:	•	• 0	• 0	• 1	3.	3
	SHOP REPAIR	TOTAL	13466	13482	13468	13408	1384	13882	1.186.0	1.3CAB	13040	13086
44 NO.	ö	698.00	21.00	54.00	11.00	14.00	2.10	54.00	3.00	17.00	7.00	275.00
45	DCT PASE EFFAIR	100.00	100.00	186.66	188.88	188.88	186.80	186.88	188.88	186.88	120.00	388.88
46	FCT CEPOT REPAIR		:	•		:	•	•	7		•	;
47 AV	42 AVERAGE BASE REPAIR CYCLE	13.88	-12	999	50.88	50016	60.45	2.64	56.81	11.56	68.81	4.6
	POT ACTIVE REPAIR	100.00	100.00	100.00	100.00	110.01	101.00	100.00	163.00	100.00	100.00	110
TO NO.	2	200						4		1		
	OF THE DADW. OF THE	900	:				•		•	7	•	21.00
100											•	
		TOTAL	13000	13540	13580	13FA0	13F88	13CAB	136A	130 88	13680	13600
NO.	OF PES	688.00	19.66	12.36	35.00	400	3.84	2	36.66	d	41.44	12.A
45	PCT EASE REFAIR	100.74	160.00	186.00	131.61	100.00	100.00		190.00	•	183.88	19: . 5:
4	PCT DEPOT PEPATE		4		4	4	1	,	4	BA	9.	1.
47 AV	AVEPAGE BASE REPAIR CYCLE	10.68	2.26	60.10	3.64	66.01	61.10	• •	.16	•	57.92	3
4.8	PCT ACTIVE REFAIR	160.40	150.00	180.00	101.01	186.88	188.84	•	101.00		100.60	106.00
64	PCT WEITE SPACE	•6.	:	03.	•	:	:	:	•	2	90.	
SO MC.	OF ITERS IN REPAIR (ECP)	57.03	1.00	40.5	•	**	1.88				12.44	4
51 NO.	. OF ITEPS BACKLCGGED (ECP)	1.60	;	:	:	•	1.			3		3
	SHCPREFAIR	TOTAL	12660	DIHER								
* ON 33	. OF PEPARABLE GENERATIONS	688.96	19.6	.;								
34	PCT BASE LEEA19	103.00	100.00									
	FCT CEPOT REPAIR	•	,	9.								
47 4 18	AVERAGE RASE REPATS CYCLE	10.83	68.12	9								
ç	FCT ACTIVE PEPAIR	199.63	106.00	:	A III							
84	PCT WHITE SPACE	3	3		-							
50 NO.	. OF ITEMS IN PERAIR (ECF)	57.00	3.6									
-												

Figure L-23 INITIAL LGAGE SIMULATION

SIMULATION
LGAGE
INITIAL
Figure L-24

	TOT DOLLA	F P LY CIRES (FOP)	TOTAL	OTHERS	AAAGE	AABos	AAEUD	ACZIG	AC220	AC236	AT 910	AF928	A1938
55 F	RATE	PERCENT	169.63	:3	100.08	130.00	188.			100.00	100.00		
	MUMBER OF	BACKORD ER-DAYS	D	8.	B.	B.	•	-	9	P	O		
	UNBER OF	NUMBER OF UNITS DEMANCED	796.00	:	10.01	6.11	1.00	1.00	;	1.11	1.00	•	
	130	CEE THE SHELE	100.00		-	10000	101-00	1000	4	1010	100.00	9	4
200		EXPRESION REFAIR	• •	D	• 6			•		-	• •	•	٠,
2	100	PEMAKES NOT SATTS.		•	42	4			3	•	4	•	إ.
- 1	NUMBER OF	CANNIBALIZATIONS	6	e		•				Ų	B		: :
2 P	NO. ITEMS	ON BACKORLER (ECP)	•	3	3	•	:	:	:	:	• •	• 73	<u>-</u>
	S 11	P P 1 Y	TOTAL	13440	13ABD	13AC#	1.3ADB	13EA	13880	1390	13040	13040	4 TC. AC
54 T	TOT DOLLAS	DOLLAR INVEST. (1000) (EOP)	9000.03	386.08	300.00	310.00	300.00	300.00	300.00	300.00	311.00	300.00	300.60
55 E	SS FILL RATE PERCENT	PERCENT	100.88	105.08	180.08	180.00	116.60	188.88	188488	188.88	186486	101.11	101 - 00
		BAC FORD ER - DAYS	9	5	٠٠	•	•	•	•	•		:	
1	NUMBER OF		786.80	21.88		11.08	16.00		56.00	3.00	17.00	ZaBB	275.1
N 40		CFF-JME-SPELF Experted memato				100	100.0	101	100.00	100.00	100.00	10.	196.
١	100		6			6			١			•	١.
£ 1	FCT	DEMANCS NOT SATIS.	9		d		. d					• •	
	NUMBER OF		0	0	0	•	:	-	٤		2	•	
- 1	NO. ITEMS		0.0		9	•	-	•					4
		2											
7 7 5	TOT DOLLA	DOLLAR THVEST (1988) (FOO)	DATO -	13555	TOPY	13589	1 STAB	13756	13648	1 36AB	13688	13686	1 360;
1		PERCENT	183.00	180.58	00.00	100		, ,			2000		
	MUMBED OF			0									
	NUMBER OF		706.00	39.00	12.36	35.68	1.86	N.A.	•	46.80	•	40.00	42.8
- 1	PCT		100.00	188.68	150.85	100.00	100.00	100.00	d	180.88			1
5.9	FOR	EXPECITED REPAIR	•	•	•	•	•	•	•	•	•	•	:
9	FCT	EREEPETICA	9.	9	4	-	-	4	4	4		4	
61	PST	DEMANDS NOT SATIS.	•	• 0	•		:	÷	Ç.	• 0	. 9	•	ت
E2 N	NUMBER OF	CANNIBALI 2ATIONS	0	9	-	-	9		4		B	-	1.4
E3 N	NO. ITEMS	Ch EACKCRDER (ECF)	3.		•	•	•	÷	:	•		.0	• ,
	7 7	S 11 F P LY	folat	13GFA	DINER	1					1		
	OT DOLLA!	TOT DOLLAR INVEST. (1936) (ECP)	90 99 . 30	360.60	0								i
ŧ	FILL RATE		102.00	166.90	9								
2 2	NUMBER OF	BACKCKOER-CAYS	706.00		• ·								
1		CEF-THE-SPELE	193.046	150000									
5	FCT	EXPECTTED REPAIR	•	C.	D	;							
60	PCT	PREEMPTION	5	0	.0								
- 1	FCI	CEMANDS NCT SATIS.	B	100	a 5								
82 N	NUMBER OF	CANNIBALIZATIONS	. •	÷.	•								

LEVEL 2

10.00

SIMULATION INITIAL LGAGE Figure L-25

shop, depending upon the AFSC. The run was made and the output results indicated a mean utilization of 19 percent, with a maximum of 51 percent. This seemed satisfactory in light of the fact that some of the work centers required a set number of people for a job that is not performed often. Also, increasing utilization by decreasing the number of people would lead to an inability to perform maintenance during peak demand times, resulting in a much Twelve AMST missions were not higher mission abort rate. accomplished, yielding a 99 percent mission Figures L-26 - to L-30 illustrate the results of overall. this run.

A final LGAGE simulation, with MSBMA values for the landing gear set to their true (original) values, was not made prior to updating of LCOM to version 3.0.

Toward the end of this effort, a study was conducted which compared manpower and AGE utilization requirements of a given scenarios for the LCOM and R&M models. The study required constructing a maintenance network which could be fully utilized by both R&M and LCOM models for comparative This network focused on the AMST landing gear purposes. maintenance with flightline AGE useage but The LCOM simulation RMLCOM was run using the maintenance. network and both manpower and support equipment data were output from the post processor programs. A short program, called TIMETOIL, was written to extract AGE usage time against each type of AGE from the voluminous data produced by LCOM. In order to extract flightline AGE data from the R&M model, two changes to the model program were necessary. The first change involved (a) calculating usage time on the flightline in a manner similar to that used for calculating personnel utilization time and (b) formatting a presentation for the output data. This change is now designed to handle any flightline resource that might turn up in Extended Form ll networks.

The second change implemented in the R&M model program was specific to the effort, wherein there are different possible combinations of AGE for each of possible landing gear failures. In LCOM, this parrallelism is handled through node probabilities. Since parallelism is not within the designed scope of the R&M model, a "work around" was conceived such that additional or parallel subsystems having only flightline tasks were permitted, and each subsystem of a set was used to model an AGE combination for the given failure type. For example, a main landing gear failure required one of five AGE combinations (e.g., a D-60 or D-60 and an NF-2) for flightline maintenance. Thus five main

OPERATIONS	Taret	1 1 U V	028139	PE1272	PEIPNI	PFS9an	PESZYR	PESTRA	PE 32YP	PESPMI	PE
1 NUMBER OF PISSIONS RECUESTED	3335.07	3121.11	43.32	4.50	7.44	16.88	2.33	5. us	2.45	2.46	3: 11.
- 1	3.32.3.1.4	311 6.04.11		8 2 4 8	7.25	90 4 4	2.35	2,00	5.0	•	
PEFCENT ACCOMPLISHED	93.64	89.62	100.65	108.60	100-04	149.00	100.38	193.00	100	103.80	•
KUPPER OF SCHILLS REGIENTED	3335.23	1		•	7.00	44.88	2404	. B. D. B.	2-01		12.00
DANSE ACCORDITIONS	1127.10	31.6.6	03.00	8 9 8	7.80		9.30	5. 80	200	2	
FIRST ACCORDINGHED	49.66	54.62	10.1.46	130.00	100.10	101	36.44	427.64			
10.4.684	-		١.					-	4		1
1000	• •	٠,	• :	• •	•	•	•	•	• •	•	•
4	•	44	4	4	4		44	4	9	4	-
MUNDA	2	٥	• • • • • • • • • • • • • • • • • • • •	•	÷	:	•	ċ	ذ	-	2
TA. NUMBER OF ATTAINERS	4		•	•		**	4		1	•	
IS NUMBER OF PAN REPAINS		,	250	•	é	*	٤	á	3	ė	
MURBS, OF A10	4	4.1			0			3			ä
			276	00000	20.00	00000	2000	201.00			
	7716	FE467R		766254	PESTIR	. PE 8980	PE7930	OF MEN			
AUTSEC AT TINCE PEGUNDEN	4445	1144	9000	18.024	384	22486	12000	4	4		
	3323.30	3			20.66	18.22	12.60	•			
	49.66	Thent	282581	128.1	28441	4777	2111	4			
E NUMBER OF SCRIPES RECUESTED	3335,13	47.4	4.4	19.61	3.80	22.00	12.00	•			
NUMBER ACCOMPLISHED	3323au	4aLB	4.36	18.11	3.10	22.BA	12446	. *			
B FEPCENT FULDMPLISHED	49.66	23. 121	150.88	100.68	100.00	200,00	180.00	:			
NUME C. LF	9	in a	4	BA	BA		-	4		!	
HO MANAGE		• ټ		•	•	•	• •				
2	d ii	E.	3.0	44	4	1	43	*			
HUMBE, OF		• •	•	ë	:	:		6			
- 1	30	- Ra	2	4	1	4	5.	9.0			
6 NUMBE OF NR FEEFIS (OTAL)	0.		.		•	:	•				
* 7 8 G B 3 8 8	TOTAL	insi	14.1351	x062	X MD3	XH27H	XMF2	X7.130.6	MORFEX	OTHER	
3 NUMBE: OF ALRCHAPT FUTH. (LCP)	1.52.	B 2028	5.86	30.86	7.30		15.00	12.50	9.90	:	
MUMBEL OF ATHEMST 1-1.3 YS AVAIL	23760 45.1	1.6			1260.06		2786.40	_	1480.00	6	
FILECKTES (INCL ALEAT)	2.74		.01		1	1 _	1	00	и.	ő	
16 F.T UNSCREG RAINTERANCE	4.15	AAA	3	•	•			7.	2	4	
POT SCHED MAINTENANCE	1.68	3.45	.25	1.53	420	.58	. 9 J	.18	61.	5	
19 F.1 MCdi	9	1.		1		, i	G.	9	•	•	
19 PCT RISSIGN MAIN STATUS	4:0	51.5	.02	.02	.01	18.	.81	18.	11.	•	
Zu C. SERVILE + MAILING	3.42	7.14	4	1.99	.00	.67	1.18	O.	e e		
21 FUT OFERSTONELLY READY	91.54	M1.66	54.68	36.44	99.75		97.67	92.80	64.65		
AVG. 10. FOST SCATTE 11ML (HPS)	7.63	£ . 1.5	09.4	43.32	4.18	29.77	35.74	4.62	23.63	7	
23 4VG. NC. CF SOF IFS/ 4/L /CDY	44.	.33	. 11	20	16.	10.	\$ 0.0	190	100		
FLYING HOLRS	15£11,75	15545.24	-	16.51	1.68	9	20.00	2.20	1.20		
17 MUMBER OF FOF TASKS FLOWN	.,					٥	:		.5	,;	

Figure L-26 CONSTRAINED LGAGE SIMULATION

2 2	FRONEL	YCLAI	2325	12385	123X1	62330	42224	42350	42354	427X4	431X2	431Y2
27 MANFOLES AVALLABLE	11LABLE (194)	1166.40	57.66	172.80	14.48	43.20	28.80	43.26	87.68	16.67	172.60	26.60
PERCENT		13,20	5145	30.92	116	27.25	26456	44.62	48.12		1.99	30.6
MANHO	(114)	223.94	23.41	6.78	. 12	11.99	7.65	19.28	27.72	•	6.49	1.65
30 PUI - UI	PLI UNSCHED MAINTENANCE	55.71	10.66	44.58	4	180.86	196.80	186.84	183.88		494	23.86
21 PLT SC	SCHEE MAINTENANCE	+4.23	96.59	55.32	:	:	:	•		•	91.36	77.86
	HEN LEHANLED	2356400	288.30	2893.56	\$	246.48	359.86	1386.88	1029.68	4	1795.01	28.46
FCT		74.63	65.56	99.26	111.63	51.65	22.36	\$6.37	57.12	•	82.62	91.00
134		•	4	1		4	4	4			4	4
FC3	FROV. BY EXPELITE	4 • 45	•	•	:	•	÷	•	ė	•		<u>:</u>
37. Pul Ps	PEDY. BY FREEMPTICH	1441	211	4		4	1	F.	4	4	٩	4
36 FOT CE	CEMANES NCT SATIS.	13.26	12.32	.72	•	40.35	77.64	48.63	15.38	;	17.38	22.65
	OVERTIME PANHOLL'S USED (12:3)	7-11	-51		1	1125	591	4441	122	4	4	7
40 SIPULATEC NE	SIPULATED NE PER FLYING HOUR	1.43	•19	40.	•	•	••	.12	•	•	*:	1 2 2
Or Ou	PERSCHNEL	TOTAL	32.124	4313k	63131	12150	ASTEA	13151	631.35	53159	63154	53155
27 MANHOUPS SVA	MANHOUPS AVAILABLE (1920)	1166.45	14.48	14.40	28.86	20.00	14.46	172.83	24.60	24.49	28.80	26.6
	LIZATIOA	19.20	59	10.61	30.67	17.63	10.18	45.49	121	2433	22.62	22.42
	5 (196)	223.94	61.	1.50	4.66	16.4	2.93	72.73		45.0	6.57	6.60
	FUT HASCHED METATENANCE	55.71	106.68	168.00	100.88	188.80	188.08	27.58	102.84	186.02	164.08	121-10
	SCHED RAINTENANCE	44.24	B.	:	3	•	4	72.42	•			-
MUMBER OF	- 4	235644.14	2000	145.45	533.68	X51428	154.0410	4617.85	3.88	15.63	326.00	324.10
53	ALAILFBLE (FPIME)	74.83	193.06	69.81	59.47	53.96	71.52	73.43	164.00	00.30	73.31	73.5
35 FUL A	ANATIABLE ISUBST. 1	9.	4	4	4	4	4	4	4	4	ē	4
FOI	FROV. BY EXPEDITE	4.45			<u>:</u>	:	:	2.86		;		*
- 1	PEGM. AY PREFMETICH	1001		4	4	4	4	3054	4 7	4	AA	97
6. C	CEMANDS NCT SATIS.	19.26	٠	30.15	40.53	*6.44	28148	13.61	•	26. 40 t	56.69	26.44
	DLE	7411	411	- 12	48	25	284	A		100	484	٦
4C SIPULATED PP	PEF FLYING HOUK	1.43	15.	.31	90,	=		. 47	•		*6.	•
P F 3	ASCHBEL	rute	EB 3 X E	OTHER								
	HIMPOLFS AVAILABLE (107)	1166.43	172.64	3								
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21 FOT 30	SCHEL PEINTENANCE	44.23	540.64									
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Figure L-27 CONSTRAINED LGAGE SIMULATION

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Figure L-28 CONSTRAINED LGAGE SIMULATION

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Figure L-30 CONSTRAINED LGAGE SIMULATION

landing gear subsystems were placed in the network. The total task time for each task type (e.g., troubleshoot or remove and replace) was then proportionately subdivided among the five subsystems, based upon the likelihood of a given AGE combination. The use of different support equipment combinations was considered in the shop tasks for this scenario, so that only one of each subsystem type was required for those tasks.

Results of this comparison study are presented in Table L-2 and Table L-3. For the AGE demand, the percent difference in overall usage time between the two models is less that 4 percent of the total. The primary reason for the differences in the two models appears to be due to the random nature of the Logistics Composite Model in failure selection and the probability distribution of the maintenance task times.

Table L-2

COMPARISON OF AGE USAGE (FLIGHTLINE)

BETWEEN LCOM AND R&M

	upa /wr			Δ % =
	HRS/KE	'H		LCOM-R&M
Unit	LCOM	R&M	(Δ) (HRS)	R&M
AJ35T	114.92	137.87	-22.95	-16.65
D60	592.45	609.87	-17.42	-2.86
MD3	27.42	25.18	2.24	8.90
M27M	53.51	51.41	2.10	4.08
NF2	237.50	250.90	-13.40	-5.34
TJ30E	53.51	51.41	2.10	4.08
TJ30 N	53.51	51.41	2.10	4.08
TOTALS	1132.82	1178.05	-45.23	-3.84

The results shown in Table L-3 for manpower utilization in maintenance manhours per thousand flight hours (MMH/KFH) show approximately a 5 percent overall difference between the models. The reasons are again primarily those discussed above. The two columns of R&M output data are listed to illustrate the slight changes in workhour needs due to differences between the model which incorporates AGE usage and that which also handles the different AGE combinations. One column lists the number of personnel

COMPARISON OF MANPOWER UTILIZATION BETWEEN LCOM AND R&M

Table L-3

		MI	MH/KFH			
		LCON 1	R&M (Modified	R&M	Δ % = 1 - 3	Mean # Personnel/
	AFSC		for AGE) 2	(Orig.) 3	3	Work Center
1	42330	329.38	348.08	348.08	-5.37	6
2	42334	222.56	257.11	230.19	-3.31	5
3	42350	472.20	524.11	508.96	-7.22	9
4	42354	583.54	626.65	626.65	-6.88	11
5	4313R	0.27	25.32	12.41	-97.82	1
6	4313W	43.38	42.15	42.15	2.93	1
7	43131	244.41	249.23	248.41	-1.61	5
8	4315R	153.25	149.70	147.16	4.14	3
9	4315W	57.86	55.27	55.27	4.68	2
*10	53151	931.67	609.87	609.87	52.77	(16) 11
11	53135	. 123	.165	.165	-25.65	1
12	53150	3.35	3.41	3.41	-1.87	1
13	53154	73.27	84.08	84.08	-12.86	2
14	53155	73.39	84.24	84.24	-12.88	2
Totals		3188.65	3059.39	3001.05	6.25	(65) 60

*includes inspection, preflight and post flight tasks in LCOM which are not in R&M - approx. 338.91 MMH/KFH are used - removing this, yields:

10	43151	592.76	609.87		-2.81
		and overall,			
Total	s	2849.74	3059.39	3001.05	-5.04

needed for each work center. The total, 60, represents only direct labor personnel performing landing gear maintenance within the scope of the scenario.

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APPENDIX M SAMPLE TRAINING PLAN

C-141 AIRCRAFT BRAKE AND WHEEL REMOVAL

INTRODUCTION

This training plan presents the content, instructional components, and instructional delivery system appropriate for task-oriented training related to brake and wheel removal/replacement on the C-141 aircraft. The entire training plan was developed employing a modified version of the instructional system development (ISD) process described in Air Force Manual 50-2. The training outlined by this plan is complemented by a task-oriented technical manual which covers the same maintenance operation.

RELATIONSHIP BETWEEN THE ISD PROCESS AND TRAINING PLAN DEVELOPMENT

The total ISD process includes the actual development, implementation, and evaluation/revision of a given training program. Thus, the development of a training plan constitutes only a portion of the ISD process. That portion, commonly termed "training program design," is depicted in the nine-step model presented in Figure M-1. Each of the nine steps and related procedures is discussed in the next section.

STEPS IN TRAINING PLAN DEVELOPMENT

The first step in training plan development is to select the specific task areas for which training will be required. For the C-141 wheel and brake removal/replacement, task areas were identified via an integrated task analysis of an actual C-141 wheel and brake assembly (for this training plan, a modified version of the integrated task analysis procedure described in AFHRL-TR-73-43 was employed). The integrated task analysis also identified and documented each

FIGURE M-1

of the discrete tasks involved in the removal/replacement operation.

The integrated task analysis of the C-141 wheel and brake removal/replacement revealed 10 task areas. These 10 areas subsumed each of the discrete tasks. Pictorially, the task areas, as well as their relationship to both the removal/replacement operation, and the discrete tasks are shown in Figure M-2.

The second step in training plan development is to categorize each of the task areas into an instructional area. Basically, and instructional area is defined as a group of task areas which, for training purposes, are closely related. For the C-141 brake and wheel removal/replacement, the instructional areas and their related task areas are shown in Figure M-3.

The third step is to add additional instructional areas as appropriate. Additional instructional areas are sometimes required either to complete the overall instructional sequence (e.g., introduction, summary) or to achieve the objectives of task-oriented training (e.g., a special skill required for a task area). For this training plan, three instructional areas were added. The areas and a rationale for the addition of each are as follows:

- Orientation to Brake and Wheel Removal/Replacement This instructional area was added to complete the overall instructional sequence. It will be used to introduce the entire brake and wheel removal/replacement operation to the students. In addition, it will provide an opportunity to describe the process of instruction to be used during the entire training program.
- 2. Use of Special Tools for Brake and Wheel Removal/Replacement This area was added to achieve the objectives of task-oriented training. It assures that students can both recognize and use those special tools involved in C-141 brake and wheel removal/replacement.
- 3. Wheel and Brake Removal/Replacement on the C-141 Aircraft This instructional area was added to complete the overall instructional sequence. It assures students an opportunity to demonstrate skill mastery on the actual C-141 aircraft.

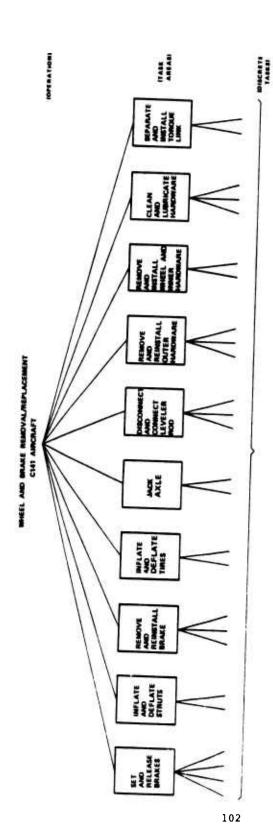


FIGURE M-2

RELATIONSHIP OF C141 BRAKE AND WHEEL REMOVALMEPLACEMENT TASK AMEAS TO BOTH THE OVERALL OPERATION AND THE DISCRETE TASKS

INSTRUCTIONAL AREA

RELATED TASK AREAS

Preparing for C-141 Brake and Wheel Removal/Replacement

- * set and release brakes
- * inflate and deflate struts
- * jack axle

Removing and Replacing C-141 Wheel

- * inflate and deflate tires
- * disconnect and connect leveler rod
- * remove and install outer hardware
- * remove and install wheel&inner hdwr.
- * clean and lubricate hardware
 (partial)

Removing and Replacing C-141 Brake

- * remove and install brake
- * separate and install torque link
- * clean & lubricate hardware (partial'

Figure M-3

Instructional Areas and Related Task Areas

The fourth step in training plan development is to construct one or more performance objectives for each instructional area (See Figure M-4). Briefly, there are two general purposes fo a performance objective (Vaughan, 1977). The first is to communicate to students exactly what is to be learned. The second is to communicate to instructional designers exactly what must be taught. In order to meet both of these purposes, the performance objectives for this training plan were constructed to reflect the criteria outlined by (Mager 1962).

In addition to the Mager criteria, each of the performance objectives was constructed to reflect the application level of Bloom's Taxonomy of Educational Objectives (Bloom et al., 1971). Learning at the level of application will help assure that over job-related performance, rather than simply information recall, is emphasized both in the development and conduct of training. Thus, the training will be better matched to the requirements of the job. The performance objectives for the entire sequence of instruction related to C-141 wheel and brake removal/replacement are presented in Appendix N.

The fifth step in training plan development is to conduct a learning analysis of each performance objective. Briefly, the purpose of a learning analysis is to both state and sequence each of the prerequisite learning (termed enabling objectives) required for successful attainment of a performance objective (Gagne, 1970). It is the enabling objectives which provide the specific focus for both the development of instructional materials and assessments.

As with the construction of performance objectives the content of each learning analysis reflects the levels of learning within Bloom's Taxonomy. This assures that the list of enabling objectives within each analysis is complete (reflects all of the prerequisite of learning required the performance objective well as correctly sequenced (in respect to the re complexity of Bloom The learning learning types). yses for all of the performance objectives within the C-141 brake and wheel removal/replacement training plan are presented in Appendix O.

The sixth step in training plan development is to select the general method by which training will be delivered. Any single method (e.g., computer assisted instruction, teacher led instruction, small group instruction) or array of methods could be selected. Normally, this methodological decision is made based upon the type and complexity of the

INSTRUCTIONAL AREA	SOURCE OF INSTRUCTIONAL AREA	PERFORMANCE OBJECTIVE NUMBER
Orientation to C-141 Brake and Wheel Removal/Replacement	additional (introduction)	1
Use of Special Tools For C-141 Brake and Wheel Removal/Replacement	additional (special skill)	2
Preparing For C-141 Brake and Wheel Kemoval/Replacement	integrated task analysis	<pre>3 (strut inflation/</pre>
Removing and Replacing C-141 Wheel	integrated task analysis	6
Removing and Replacing C-141 Brace	integrated task analysis	7
Wheel and Brake Removal/Replacement on the C-141 Aircraft	additional (real world skill demonstration)	8

Figure M-4

C-141 Wheel and Brake Removal/Replacement - Instructional Areas and Related Performance Objectives

performance objectives (and related learning analyses), the ability of the learners, and the environment for training.

For this training plan, the general method of instruction" was selected. This choice was made based upon a number of considerations. First, the complexity of the performance and enabling objectives is not of a level which would preclude presentation via a self-instructional format. Second, recent trends in training include a movement toward valid and predictable instruction which is exportable (i.e., capable of replication at a variety of locations without special facilities or personnel). By definition, validity and predictability are consistently achievable only via highly structured self-instructional approaches Third, the concept of self-instruction will learning. permit students to spend as much or as little study time as required in order to achieve the goal of mastery learning.

The selection of self-instruction is not intended to remove, or reduce the importance of, the instructor within a training program. Rather, self-instruction permits many of the lower level types of learning to be mastered by a student working independently. Instructional personnel are thus able to devote their expertise to the more crucial aspects of training; e.g., complex skill development, individual student supervision, individual student assessment, and remediation.

The seventh step in training plan development is to select the primary instructional delivery technique which will adequately carry out the general method of instruction. For this training plan, "instructional modules" were chosen as the primary delivery technique. The choice was made because instructional modules are useable by a wide range of students who possess varying abilities to learn, speed in learning, and levels of interest. Further, a given instructional module may include a variety of different media forms reflecting various student interests (a more thorough discussion of instructional module; in general and for this training plan may be found in Appendix P).

The eighth step is to select an assessment type for each performance and enabling objective. The assessment type selected for the performance objective will indicate the manner in which student performance (for purposes of grading) will be judged. The performance objective statement normally indicates the assessment type to be used.

The assessment types selected for the enabling objectives will indicate the format of diagnostic tests (when given before instruction) and/or self-tests (when given following instruction). However, enabling objective assessment items are not used for grading purposes.

The assessment type chosen for a given enabling objective is based upon two constraints. First, that the assessment type can adequately measure the behavior specified in the enabling objective. Second, that any restrictions imposed by the instructional method (in this case self-instruction) or the instructional delivery technique (in this case modules) can be reflected. The enabling objectives and a preliminary choice of assessment type for each presented in Appendix Q.

The ninth and final step in training plan development is to select a media type for each enabling objective.*

As with assessment types, media types are selected in respect to the behavior specified within the objective, as well as, by any restrictions imposed by either the method of instruction or instructional delivery technique. In addition, media type is often a direct product of the abilities and interests of the students who will undertake the instruction.

For the training plan under discussion, a single tentative media type has been selected for each enabling objective. Alternative and/or additional media types may later be chosen in order to satisfy the different learning abilities and interests of different students. The tentative media type selected for each enabling objective is presented in Appendix Q.

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^{* (}Note: Media types are not selected for a performance objective because performance objective is not directly taught. Rather, the behavior specified in the performance objective is the result of the attainment of related enabling objectives.

VALIDATION AND USE OF A TRAINING PLAN

The preceding paragraphs described that portion of the ISD process required to complete the development of a training plan for brake and wheel removal/replacement on the C-141 aircraft. Those paragraphs, as well as the content of Appendices N to Q and constitute the training plan. This training plan for the brake and wheel removal/replacement operation is considered preliminary and should undergo review and revision by three groups of individuals:

- 1. Instructional designers who can analyze the contents of the plan in respect to accepted standards of instructional design.
- Equipment user's who can analyze the contents of the plan in respect to the maintenance needs of Air Force work units.
- Equipment builders who can analyze the contents of the plan in respect to the maintenance requirements of the equipment under consideration.

This validation process would continue until an acceptable degree of consensus on the contents of the training is obtained. At this point, the training plan would be deemed "final."

Upon acceptance of a final training plan, the following ISD activities would be undertaken:

- 1. Development of the assessment items.
- 2. Development of the instructional materials.
- 3. Packaging of instructional materials and assessments into self-instructional modules.
- 4. Training for the instructors vis-a-vis the use of self-instructional modules as well as the supervision and evaluation of student performance in a self-instructional setting.
- 5. Design and development of required equipment mockups.

- 6. Field test and revision of the instruction in order to achieve predictability and consistency.
- 7. Deployment of training.

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APPENDIX N TRAINING PLAN PERFORMANCE OBJECTIVES C141 WHEEL AND BRAKE REMOVAL/REPLACEMENT

PERFORMANCE OBJECTIVE 1:

Given 10 multiple choice items, the student will be able

- a. to identify the tasks and task sequence involved in the removal/replacement of a C141 wheel and brake assembly and
- b. to recognize situations in which removal/replacement of the wheel and brake assembly is necessary.

To successfully achieve this objective, the student must correctly answer nine of the test items.

PERFORMANCE OBJECTIVE 2:

Given the five special tools required for brake and wheel removal/replacement on the Cl41 aircraft and mockups of fasteners requiring the tools, the student will

- a. select the tool required by each fastener and
- b. use each tool to loosen and tighten the related fastener.

To satisfy this objective, the student must follow the tool use and safety procedures outlined in the module.

PERFORMANCE OBJECTIVE 3:

Given mockups of a high pressure nitrogen servicing cart and of a given Cl41 main landing gear strut, a height to which the strut is to be raised, and a Cl41 strut inflation/deflation checklist, the student will be able to inflate and deflate the strut mockup. To successfully achieve this objective, the student must accomplish the following steps:

- Determine if strut inflation/deflation is required.
- 2. Set up, operate, and shut down the servicing cart.
- 3. Ensure that maximum strut extension limits are not exceeded.
- 4. Ensure that maximum strut servicing limits are not exceeded.
- 5. Ensure that maximum permitted differential in strut heights are not exceeded.
- 6. Carry out each of the strut inflation/deflation procedures indicated in the module.
- 7. Carry out each of the related safety procedures indicated in the module.

PERFORMANCE OBJECTIVE 4:

Given mockups of a C141 main landing gear strut and aircraft jack, and a C141 axle jacking checklist, the student will be able to raise and lower the strut (aircraft). To successfully achieve the objective the student must accomplish each of the following steps:

- 1. Inspect the jack prior to use.
- 2. Correctly position the jack.
- 3. Raise the wheel approximately one inch.

- 4. Carry out each of the axle jacking procedures indicated in the module.
- 5. Carry out each of the related safety procedures indicated in the module.

PERFORMANCE OBJECTIVE 5:

Given a brake number, mockups of C141 rudder pedals, hydraulic system control panel, and a C141 brake setting/releasing checklist, the student will be able to set and release the brakes. To successfully achieve this objective, the student must accomplish the following steps:

- 1. Pressurize the correct hydraulic system for the given brake.
- 2. Check that the brakes are actually set.
- 3. Carry out each of the brake setting/releasing procedures indicated in the module.
- 4. Carry out each of the related safety procedures indicated in the module.

PERFORMANCE OBJECTIVE 6:

Given mockups of a C141 axle, wheel and related inner and outer hardware incorporating one or more defective parts, and a C141 wheel removal/replacement checklist, the student will be able to remove and replace the wheel/related hardware. To successfully achieve this objective, the student must accomplish the following steps:

- Inspect all required hardware components and identify/replace and which are defective.
- 2. Correctly lubricate all required hardware components.
- Use all tools correctly.

- 4. Torque wheel to proper tolerances.
- 5. Verbally indicate cautions and carry them out.
- 6. Carry out each of the wheel removal/replacement procedures indicated in the module.
- 7. Carry out each of the related safety procedures indicated in the module.

PERFORMANCE OBJECTIVE 7:

Given mockups of a Cl4l axle, brake assembly, strut, and torque link incorporating one or more defective parts, a brake removal and replacement checklist, and the instructor acting as a hydraulic specialist, the student will be able to remove and replace the Cl4l brake assembly/related hardware and recognize/replace all defective parts. To successfully achieve this objective, the student must accomplish the following steps:

- 1. Request the hydraulic specialist to depressurize the hydraulic system and cap/plug the cap/filler.
- Use all tools correctly.
- Inspect all required hardware components and identify/replace any which are defective.
- 4. Correctly lubricate all required hardware components.
- 5. Verbally indicate cautions and carry them out.
- 6. Carry out each of the brake removal/replacement procedures indicated in the module.
- 7. Carry out each of the related safety procedures indicated in the module.

PERFORMANCE OBJECTIVE 8:

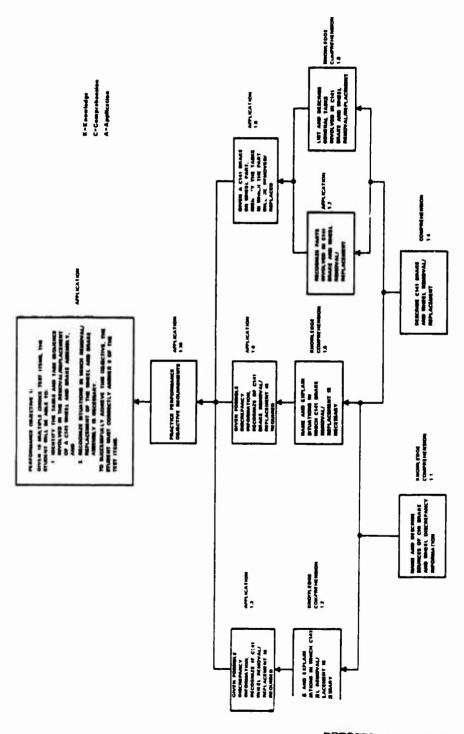
Given an actual Cl41 aircraft, and related Cl41 job guide, the student will be able to remove and replace both the wheel and brake assembly. To successfully achieve the objective, the student must accomplish the following steps:

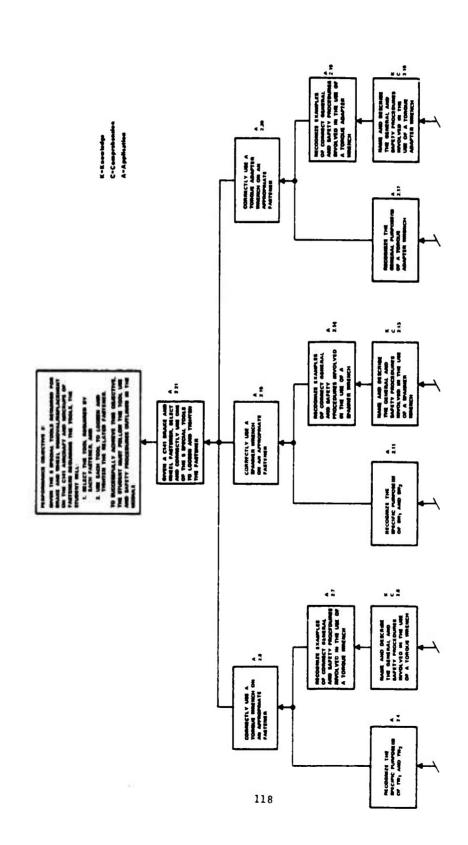
- 1. Inspect all hardware components and identify/replace any which are defective.
- 2. Lubricate all required hardware components.
- 3. Verbally indicate all cautions and carry them out.
- 4. Carry out the aircraft procedures for the task as indicated in the module.
- 5. Carry out each of the related safety procedures as indicated in the module.

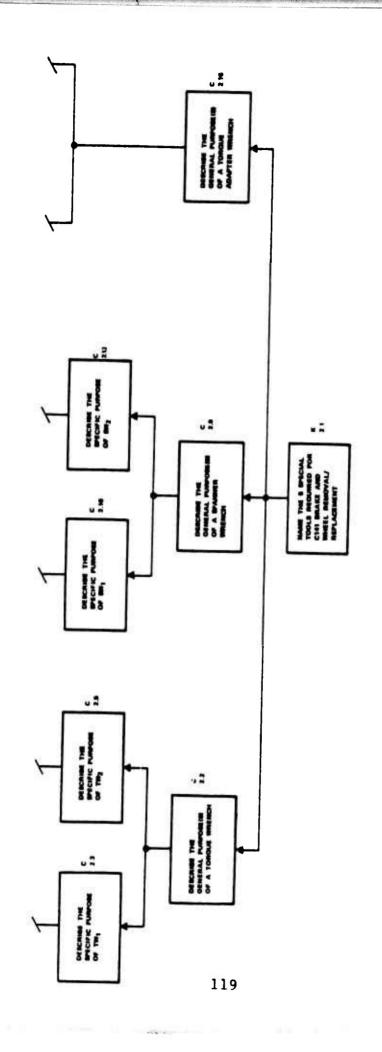
APPENDIX O

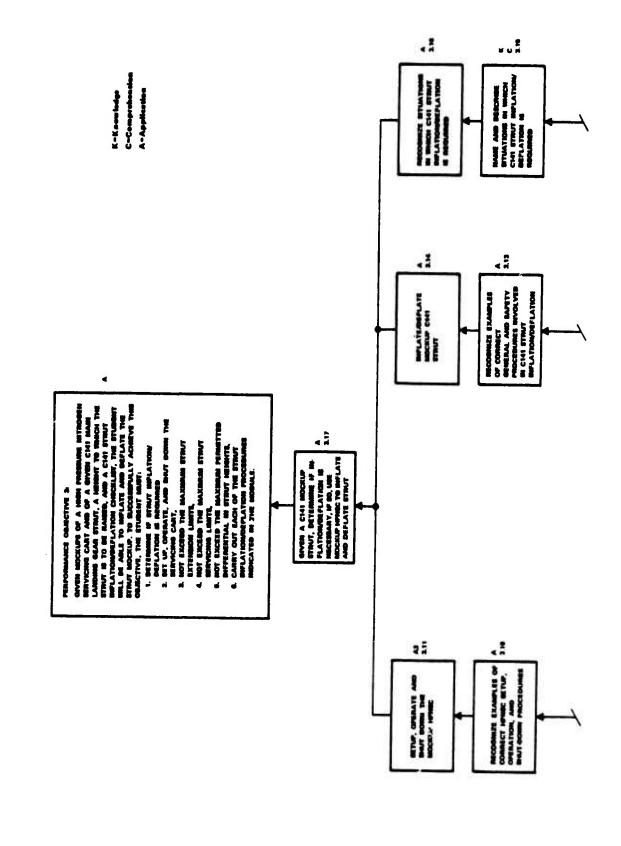
TRAINING PLAN LEARNING ANALYSES

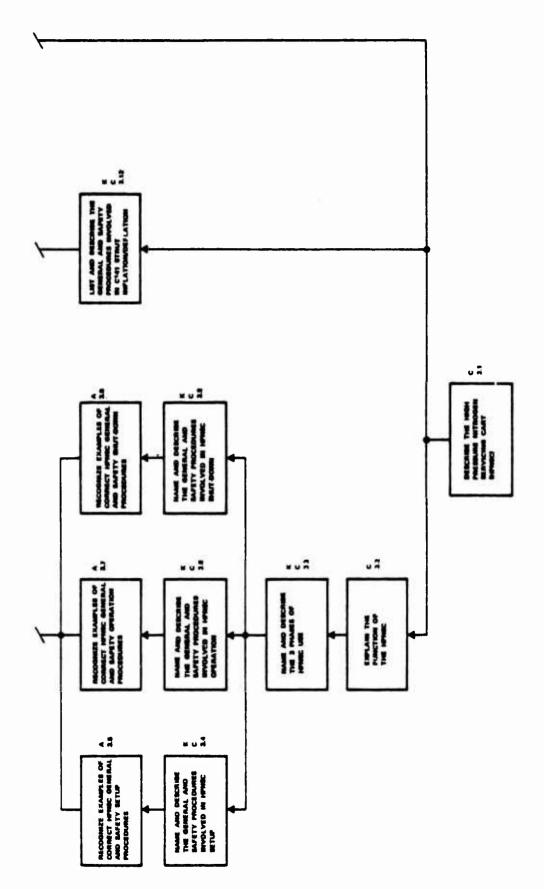
C141 WHEEL AND BRAKE REMOVAL/REPLACEMENT

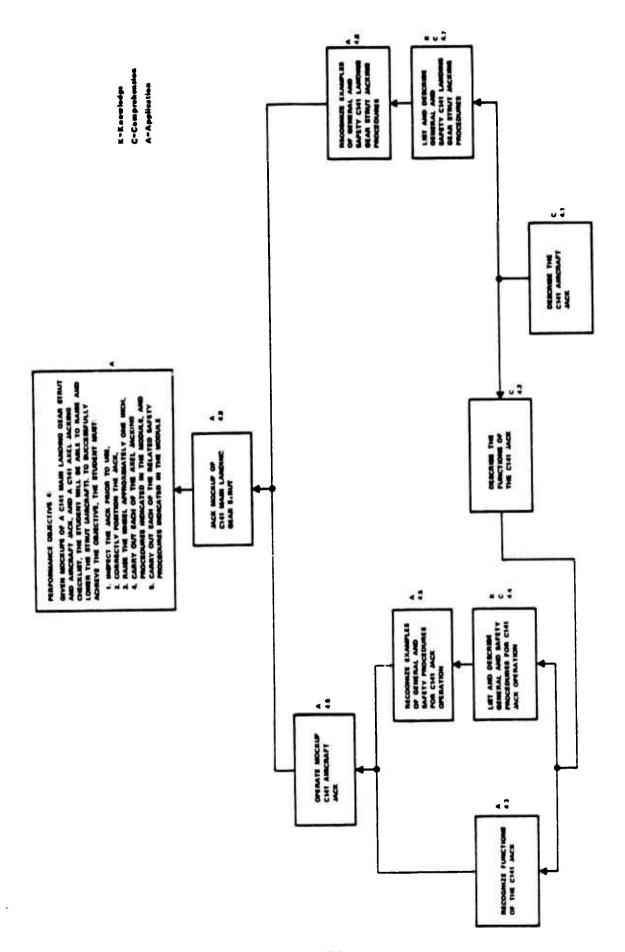




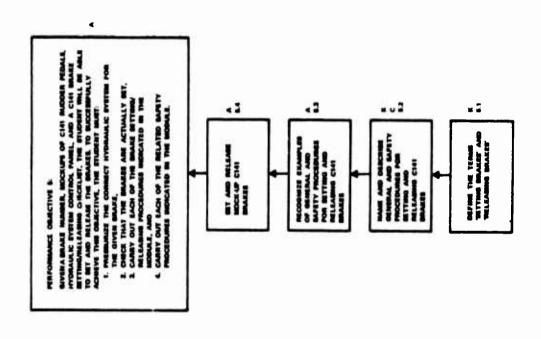


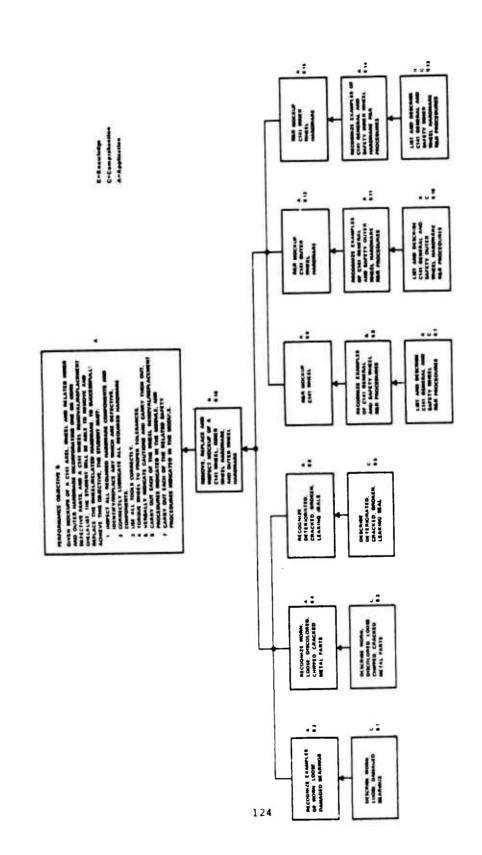


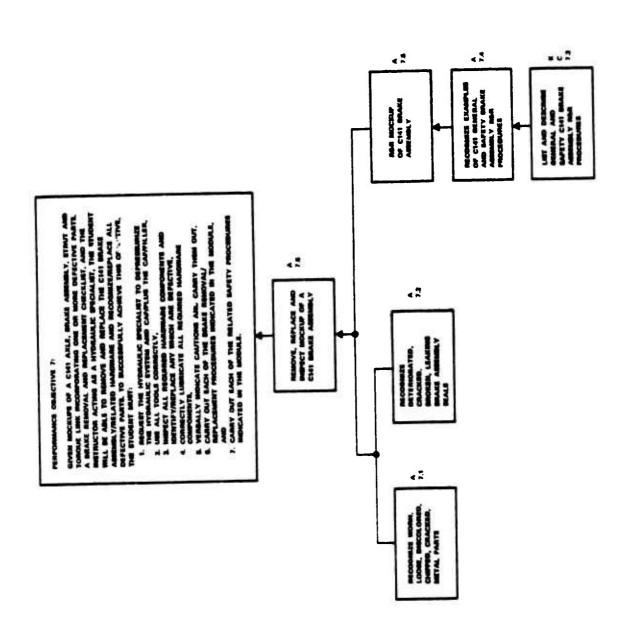


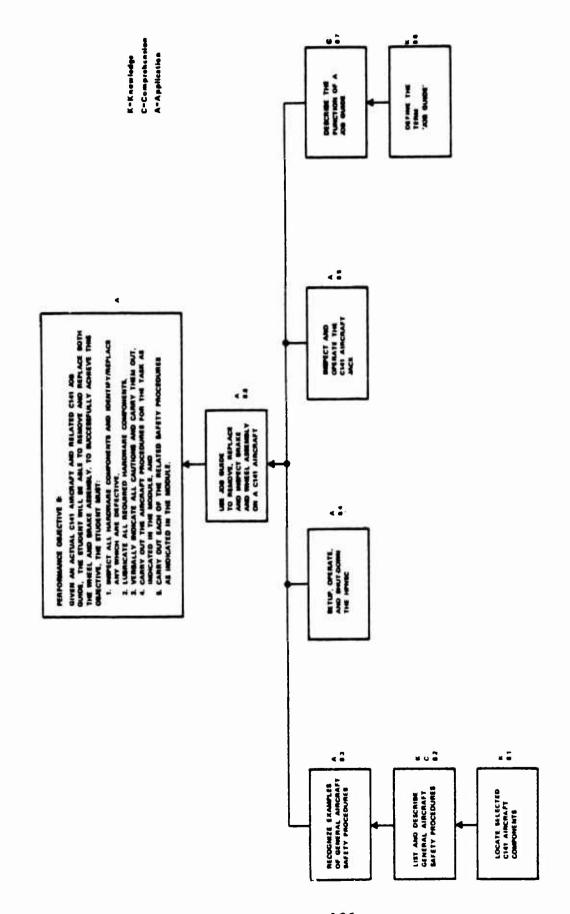


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APPENDIX P DESCRIPTION OF MODULARIZED INSTRUCTION

OVERVIEW OF MODULES

A module is an instructional package dealing with a single performance objective. It is an attempt to individualize learning by enabling a trainee to master one unit of instruction before moving to another. The learning experiences within a module are presented in a selfinstructional format which permits the trainee to control the rate and intensity of study. Since the module package may involve materials which are portable, the trainee can take it to the library, a study carrel or living quarters. The length of a module may vary from only a few minutes of trainee time to several hours. Modules may be used individually or combined to form sequences of learning devoted to a specific overall outcome (e.g., a particular job skill).

CONTENT OF A MODULE

Modules may vary dramatically in terms of content. Some modules provide only a prescription in the sense that a trainee is told where to obtain and how to use such items as instructional materials and assessments. Conversely, other modules may be totally self-contained in that they would contain everything a trainee required in order to achieve a given performance objective. For the modules related to C-141 brake and wheel removal/ replacement, the following general content is proposed:

- 1. An introduction including the statement of the performance objective.
- 2. A diagnostic test related to each of the enabling objectives.
- 3. An answer key to the diagnostic test, including a relationship chart matching incorrectly answered diagnostic test items with appropriate instruction.

- 4. All instruction required for each enabling objective (other than that limited amount of instruction requiring either instructor delivery or supervision).
- 5. Self-tests valid for the instruction within the module.
- 6. Directions on how and where the module post-test will be administered.

STUDENT USE OF AN INSTRUCTIONAL MODULE

Based upon the above proposed content, the modules would be used in the following general manner:

- 1. Trainees would begin use of a module by reading the introduction including the performance objective.
- 2. Trainees would complete the diagnostic test.
- 3. Depending upon the results of the diagnostic test, the relationship chart would guide each trainee to the appropriate instruction.
- 4. Upon completion of each segment of instruction, trainees would complete the related self-test.
- 5. Following completion of all instruction, the trainee would take the post-test related to the module. (The post-test could be given on demand or based upon a schedule provided by the instructor.)

Most of the study related to the module could be undertaken independently at a location of the trainee's choosing. However, that instruction requiring the use of mockups or special tools would be completed within the confines of a training environment under instructor control.

ROLE OF THE INSTRUCTOR

The role of the instructor in modular instruction is that of facilitator and evaluator. Essentially, instructors have limited responsibility for delivering "stand-up" instruction. Instead, they are available to assist trainees in overcoming learning problems on a one-to-one basis. Further, instructors will supervise trainee practice of skills involving mockups or special tools as well as evaluating student completion of each post-test. Thus, in a modular situation, the responsibility for instructional delivery moves from the instructor to the modules while the instructor tends to take on a heavily diagnostic role.

APPENDIX Q PRELIMINARY MEDIA AND ASSESSMENT CHOICES

C141 BRAKE AND WHEEL REMOVAL/REPLACEMENT

Enabling Objective	Preliminary Medial Choice	Preliminary Assess- ment Choice
1.1	written instruction	short answer item
1.2	written instruction	short answer item
1.3	written instruction	alternative response item
1.4	written instruction	short answer
1.5	written instruction	short answer
1.6	written instruction	alternative response
1.7	written instruction with still pictures	alternative response with still pictures
1.8	written instruction	short answer
1.9	written instruction with still pictures	alternative response with still pictures
1.10	alternative response	alternative response
2.1	written instruction	short answer
2.2	written instruction	short answer
2.3	written instruction	short answer
2.4	written instruction	alternative response
2.5	written instruction	short answer
2.6	written instruction	short answer
2.7	written instruction with sequenced still pictures	alternative response with sequenced still pictures

2.8	practice	instructor observation
2.9	written instruction	short answer
2.10	written instruction	short answer
2.11	written instruction	alternative response
2.12	written instruction	short answer
2.13	written instruction	short answer
2.14	written instruction with sequenced still pictures.	
2.15	practice	instructor observation
2.16	written instruction	short answer
2.17	written instruction	alternative response
2.18	written instruction	short answer
2.19	written instruction with sequenced still pictures	alternative response with sequenced still pictures
2.20	practice	instructor observation
2.21	practice	instructor observation
3.1	written instruction	short answer
3.2	written instruction	short answer
3.3	written instruction	short answer
3.4	written instruction	short answer
3.5	written instruction with still pictures	alternative response with still pictures
3.6	written instruction	short answer
3.7	written instruction	alternative response

3.8	with still pictures written instruction	with still pictures short answer
3.9	written instruction with still pictures	alternative response with still pictures
3.10	written instruction with still pictures	alternative response with still pictures
3.11	practice	instructor observation
3.12	written instruction	short answer
3.13	written instruction with still pictures	alternative response with still pictures
3.14	practice	instructor observation
3.15	written instruction	short answer
3.16	written instruction with still pictures	alternative response with still pictures
3.17	practice	instruction observation
3.17	practice written instruction	
		observation
4.1	written instruction	observation short answer
4.1	written instruction written instruction	observation short answer short answer
4.1 4.2 4.3	written instruction written instruction written instruction written instruction	observation short answer short answer short answer short answer alternative response
4.1 4.2 4.3 4.4	written instruction written instruction written instruction written instruction written instruction	observation short answer short answer short answer short answer alternative response
4.1 4.2 4.3 4.4 4.5	written instruction written instruction written instruction written instruction written instruction written instruction with still pictures	observation short answer short answer short answer short answer alternative response with still pictures instructor
4.1 4.2 4.3 4.4 4.5	written instruction written instruction written instruction written instruction written instruction written instruction with still pictures practice	observation short answer short answer short answer short answer alternative response with still pictures instructor observation

5.1	written instruction	completion
5.3	written instruction with still pictures	alternative response with still pictures
5.4	practice	instructor observation
6.1	written instruction	short answer
6.2	written instruction with still pictures	alternative response with still pictures
6.3	written instruction	short answer
6.4	written instruction with still pictures	alternative response with still pictures
6.5	written instruction	still pictures
6.6	written instruction with still pictures	alternative response with still pictures
6.7	written instruction	short answer
6.8	written instruction with still pictures	alternative response with still pictures
6.9	practice	instructor observation
6.10	written instruction	short answer
6.11	written instruction with still pictures	alternative response with still pictures
6.12	practice	instructor observation
6.13	written instruction	short answer
6.14	written instruction with still pictures	alternative response with still pictures
6.15	practice	instructor observation
7.1	written instruction with still pictures	alternative response with still pictures
7.2	written instruction	alternative response

	with still pictures	with still pictures
7.3 7.4	written instruction written instruction with still pictures	short answer alternative response with still pictures
7.5	practice	instructor observation
7.6	practice	instructor observation
3.1	written instruction with still pictures	alternative response with still pictures
8.2	written instruction	short answer
8.3	written instruction with still pictures	alternative response with still pictures
8.4	practice	instructor observation
8.5	practice	instructor observation
8.6	written instruction	completion
8.7	written instruction	short answer
8.8	practice	instructor observation

APPENDIX R SAMPLE JOB GUIDE MANUAL

VE AND INSTALL MAIN LANDING GEAR LS AND BRAKES

INPUT CONDITIONS

Applicable Serial Nos: All Supplies:

Clean rags (as required) Grease, MIL-G-25760 Oil, MIL-C-6529 Cleaning solvent, P-D-680 Cap and plug kit

Personnel Required: Two

Assistant required to aid in jacking and handling wheel and brake during remove and install.

Specialist required to connect and disconnect hydraulic swivel fittings during remove and install brakes.

Technician required to perform In-Process inspection during installation.

Equipment Condition:

Aircraft parked, chocked and statically grounded per T.O. 1C-141A-2-2JG-4.

Aircraft positioned on hard, level service. Slope tolerance of 1.5 percent, zero degrees, 51-1/2 minutes.

Landing gear ground safety pins installed per T.O. 1C-141A-2-1MS-1.

Main landing gear shock strut serviced per T.O. 1C-141A-2-2JG-5.

No. 3 hydraulic system accumulators pressurized per T.O. 1C-141A-2-3JG-2.

WARNING

Observe all safety precautions in AFM127-101; T.O. 00-25A-1; T.O. 1C-141A-2-1MS-1.

CAUTION

Always jack aft axle first, or both axles simultaneously when there is a requirement to change tires on both axles on one gear.

Special Tools & Test Equipment:

Axle j Type F-2; used to jack bogie (two required if forwar it wheels are to be removed). (Alternate: Axle jacks, US; used to jack front or rear axles if both tires at tat.)

Torque wrench, calibrated to 960 inch-pounds; used to torque axle nut.

Torque wrench, calibrated to 480 inch-pounds; used to torque axle nut.

Spanner Wrench, PN 3S50034; used to remove and install axle nut.

Torque adapter Wrench, PN 7227089-10; used to check position of lockring.

Valve core Tool, PN 968RB; used to remove or install valve core. (Alternate: Equivalent)

NOTE: Under an emergency situation, or when a MLG brake leak occurs at a station where brake change capability does not exist, it is permissible to cap off a maximum of one brake per main gear for a one time flight to a station having brake change capability. Cap off brake by obtaining cap/plug adapters from crew compartment. Disconnect flex hose (4) from swivel at brake (20). Install female part of adapter to the brake swivel and flex hose (4) to male part of adapter. Using number 3 hydraulic system pressurize affected brake and check for leaks.

REMOVE WHEEL AND TIRE

NOTE: If there is not enough room between the aircraft and axle to remove the wheel, inflate both landing gear struts. (Refer to T.O. 1C-141A-2-2JG-4)

- 1. Make sure that the parking brakes are off.
- 2. Remove forward and aft chocks.
- 3. Disconnect leveler rod (37).
 - a. Remove cotter pin (36).
 - b. Remove nut (35).
 - c. Remove outer washer (34).
 - d. Free leveler rod by pulling it off of bolt.
 - e. Leave inner washer (38) in bolt and put outer washer (34) and nut (35) onto bolt for safekeeping.
 - f. Tie leveler rod to forward torque arm so it will not dangle or be damaged.
- 4. Jack axle until tire clears ground. (Refer to T.O. 1C-141A-2-2JG-4)
- 5. Set parking brake.
 - a. Depress top part of rudder pedals (40).
 - b. Pull out parking brake handle (39).
 - c. Release rudder pedals (40).

NOTE: If parking brake handle will not set in the out position, check to make sure that there is enough hydraulic pressure (Refer to T.O.

- 6. Deflate tire.
 - a. Remove air valve cover.
 - b. Use valve core tool to deflate tire until all air is out.

- c. Use valve core tool to remove valve core.
- 7. Remove outer wheel hardware.
 - a. Remove snapring (23).
 - b. Remove hubcap (5).
 - c. Remove grease retainer ring (9).
 - d. Remove felt grease seal (10).
- 8. Disconnect anti-skid detector (6).
 - a. Use wire snippers to cut and remove safety writing from screws.
 - b. Remove the three screws and place them in hubcap for safekeeping.
 - c. Place skid detector (6) inside axle (22) for safe storage.
- 9. Remove axle nut (7) and lock ring (8).
 - a. Remove cotter pin (25) from lock ring (8) and axle nut (7).
 - Use circular spanner wrench to remove axle nut(7).
 - c. Remove lock ring (8).
- 10. Clean axle nut (7) and lock ring (8) with clean rag.
- 11. Remove rest of outer wheel hardware.
 - a. Remove grease retainer ring (11).
 - b. Remove outer bearing (12), catching it in rag and keeping it wrapped in rag for protection.
- 13. With assistance, remove wheel and tire from axle.

CAUTION

In removing wheel, try to support wheel as it comes off so as to avoid damaging end of axle.

REMOVE BRAKE

NOTE: Remove wheel and tire as per this T.O.

- 1. Remove outer spacer (19) from axle.
- 2. Request hydraulic specialist to:
 - 1) Depressurize #3 hydraulic system per T.O. 1C-141A-2-3JG-1.
 - 2) Disconnect hydraulic line (4) and swivel fitting (3).
 - 3) Cap fitting and plug line.
- 3. Separate torque link (31) from brake (20).
 - a. Use screwdriver to pry lock tab on star washer (32) out of the indent on nut (33).
 - b. Use spanner wrench to loosen nut (33) 3 or 4 turns.
 - c. Tap nut (33) with hammer to free bolt (30).
 - d. Remove nut (33), star washer (32), and bolt (30).
 - e. Set torque link (31) out of the way by lifting it up and back.

CAUTION

In taking the brake off the axle care must be taken to protect the bearing surfaces on the axle.

- 4. With assistant, remove brake (20) from axle.
- 5. Use brake dolly to transport brake to service chariot.

NOTE: Sometimes the inner spacer sticks to the brake. Be sure the spacer is removed before brake leaves the area.

INSTALL BRAKE

- 1. Using solvent, clean axle (22) and spacers (19,20).
- 2. Apply a thin film of grease to axle (22).
- 3. Install inner spacer (21) on axle (22). (Make sure it goes all the way on).
- 4. Get the new brake (20) and remove any service tags and/or wire attachments.
- 5. Align the rotor and stator discs of the brake.
- 6. With assistant, install brake onto axle. (Make sure it is snug against shoulder of inner spacer (21.)
- 7. Install torque link (31).
 - a. Align torque link (31) with brake (20) so that bolt holes line up exactly.
 - b. Install bolt (30) so that flat edge of bolt head lines up with alignment bar on brake housing.

NOTE: When bolt is properly installed, the bolt will stick out the other side of the hole by about 1/32 of an inch.

c. Place star washer (32) on the end of bolt (30).

NOTE: If the star washer will not stay on end of bolt, make sure that the bolt is seated correctly and that it sticks out the right amount. If the bolt does stick out the right amount, then you can hold it in place by applying a little grease to the brake housing for the washer to stick to.

- d. Install nut (33), tightening it with the spanner wrench until all end play in the bolt is removed.
- e. Continue to tighten nut (33) with the spanner wrench (or with a punch and hammer) until a tab on the star washer (32) lines up with an indent on the nut (33).
- f. Use a screwdriver to bend a tab on the star washer (32) part way into an indent or nut (33).

- g. Use a hammer and punch to seat tab securely into the indent.
- 8. Request hydraulic specialist to:
 - Remove cap from fitting (3) and plug from line (4).
 - 2) Connect line (4) to swivel fitting (3).
 - 3) Bleed brake per T.O. 1C-141A-2-3JG-2.
 - 4) Service hydraulic system per T.O. 1C-141A-2-2JG-5.
 - 5) Check adjustment of brake and leak check brake per T.O. 1C-141A-2-12JG-6.
- 9. Set parking brake.
 - a. Depress top part of rudder pedals (40).
 - b. Pull out parking brake handle (39).
 - c. Release rudder pedals (40).
- 10. Install outer spacer (19) with the flared end toward the brake.

IN-SHOP PREPARATION OF WHEEL FOR INSTALLATION

NOTE: For wheels which have been removed and are to be reinstalled, follow steps 1-6 only. For installation of a new wheel follow steps 7-12 only.

- 1. Remove inner wheel hardware.
 - a. Remove snapring (18).
 - b. Remove grease retainer ring (17).
 - c. Remove felt seal (16).
 - d. Remove grease retainer ring (15).
 - e. Remove inner bearing (14), placing it in a rag for protection.

NOTE: Check for proper bearing part number - #PN522549.

- 2. Using cleaning solvent, clean inner and outer wheel hardware.
 - a. Clean felt seals (10, 16).
 - b. Clean retaining rings (9, 11, 15, 17).
 - c. Clean bearings (12, 14).
- Inspect felt seals, retaining rings, and bearings for wear and defects.
- 4. Lubricate felt seals (10, 16) with oil.
- 5. Repack bearings (12, 14) with grease.
- 6. Install inner wheel hardware.
 - a. Install inner bearing (14).
 - b. Install grease retainer ring (15).
 - c. Install felt seal (16).
 - d. Install grease retainer ring (17).
 - e. Install snapring (18).

YELL

NOTE: For wheels that are being reinstalled, step 6 ends the wheel preparation task. For the installation of a new wheel follow steps 7-12.

- 7. Using cleaning solvent, clean outer wheel hardware.
 - a. Clean felt seal (10).
 - b. Clean retaining rings (9, 11).
 - c. Clean bearing (12).
- 8. Inspect felt seal, retaining rings, and bearing (9, 10, 11, 12) for wear and defects.
- 9. Lubricate felt seal (10) with oil.
- 10. Repack outer bearing (12) with grease.
- 11. Remove inner wheel hardware to check for proper bearing part number #PN-522549.
 - a. Remove snapring (18).
 - b. Remove grease retainer ring (17).
 - c. Remove and discard paper protective seal.
 - d. Remove felt seal (16).
 - e. Remove grease retainer ring (15).
 - f. Remove bearing (14) and check for proper part number #PN-522549.
- 12. Reinstall inner wheel hardware.
 - a. Install bearing (14).
 - b. Install grease retainer ring (15).
 - c. Install felt seal (16).
 - d. Install grease retainer ring (17).
 - e. Install snapring (18).

INSTALL WHEEL AND TIRE

NOTE: Prepare wheel for installation as per this T.O.

NOTE: Before starting installation, check to be sure parking brakes are set.

- 1. Visually inspect wheel and tire assembly as per T.O. 1C-141A-2-1MS-1.
- Check tire pressure for proper inflation (190 +10, -5 psi).
- 3. Check to make sure that the outer spacer (19) has been installed with flared end towards the brake.
- 4. Remove inner wheel hardware to check for proper bearing part number #PN-522549.
 - Remove snapring (18).
 - b. Remove grease retainer ring (17).
 - c. Remove felt seal (16).
 - d. Remove grease retainer ring (15).
 - e. Remove bearing (14) and check for proper part number #PN-522549.
- 5. Reinstall inner wheel hardware.
 - a. Install bearing (14).
 - b. Install grease retainer ring (15).
 - c. Install felt seal (16).
 - d. Install grease retainer ring (17).
 - e. Install snapring (18).
- 6. With assistance, put wheel onto axle and engage it with brake.
 - a. Check alignment of rotors on brake.
 - b. Guide wheel onto axle.

NOTE: Lift wheel carefully so as not to damage end of axle.

- c. Keep aligning rotors as necessary while jiggling wheel all the way on.
- 7. Install outer bearing (12).
- 8. Install lock ring (8) and axle nut(7).
 - a. Install lock ring (8) with lock keys in axle lock slots.
 - b. While holding lock ring (8) in place with one hand, install axle nut (7) and hand tighten it with other hand.
- 9. Release brakes.
- 10. Using spanner wrench (round head), torque axle nut (7) to 960 inch pounds.

NOTE: While tightening axle nut, rotate wheel clockwise.

- 11. Let up on axle nut tightness by backing off to zero inch pounds.
- 12. Using spanner wrench (rectangular head), torque lock ring (8) to 600 inch pounds (in either direction) to insure that the lock keys are firmly seated in axle lock slots.

WARNING

If ring nut (8) is not properly seated, the whole wheel assembly can come off.

- 13. Using spanner wrench (round head), torque axle nut (7) to 480 inch pounds.
- 14. Continue tightening axle nut (7) until the first (of the three) locking holes you come to lines up with an indent on the lock ring (8).
- 15. Install cotter pin (25).
 - a. Insert cotter pin (25) through indent in lock ring (8) and into locking hole in axle nut (7).

- b. Secure cotter pin and flatten rounded head into lock ring indent.
- 16. Install anti-skid detector (6).
 - a. Remove anti-skid detector (6) from inside the axle (22) and place in position on axle nut (7).
 - b. Secure anti-skid detector with three screws (24).

 NOTE: Be sure screws are securely seated but do not overtighten them.
 - c. Safety wire the three screws (24) using single strand method.
- 17. Install rest of outer wheel hardware.
 - a. Install grease retainer ring (11).
 - b. Install felt seal (10).
 - c. Install grease retainer ring (9).
 - d. Install hubcap (5) with key inserted into alignment hole in wheel.
 - e. Install snapring (23).
- 18. Release parking brake.

CAUTION

Be sure to get all equipment out from under the aircraft before lowering the jacks.

NOTE: If both axles are jacked, do step 17a to both jacks at the same time.

- 19. Lower aircraft.
 - a. Slowly release pressure on jack.
 - b. When jack is clear of jack pad (38.2), remove jack.
- 20. Put forward and aft wheel chocks into place.
- 21. Install leveler rod (37).
 - a. Untie leveler rod.

- b. Remove nut (35) and outer washer (34), leaving inner washer (38) on bolt.
- c. Place end of leveler rod (37) onto bolt.
- d. Put on washer (34) and nut (35).
- e. Tighten nut (35) finger tight and then back off so that one of the indents on the nut aligns with the cotter pin hole in the bolt.
- f. Insert and secure cotter pin (36).
- 22. Deflate the struts (Refer to T.O. 1C-141A-2-2JG-4).

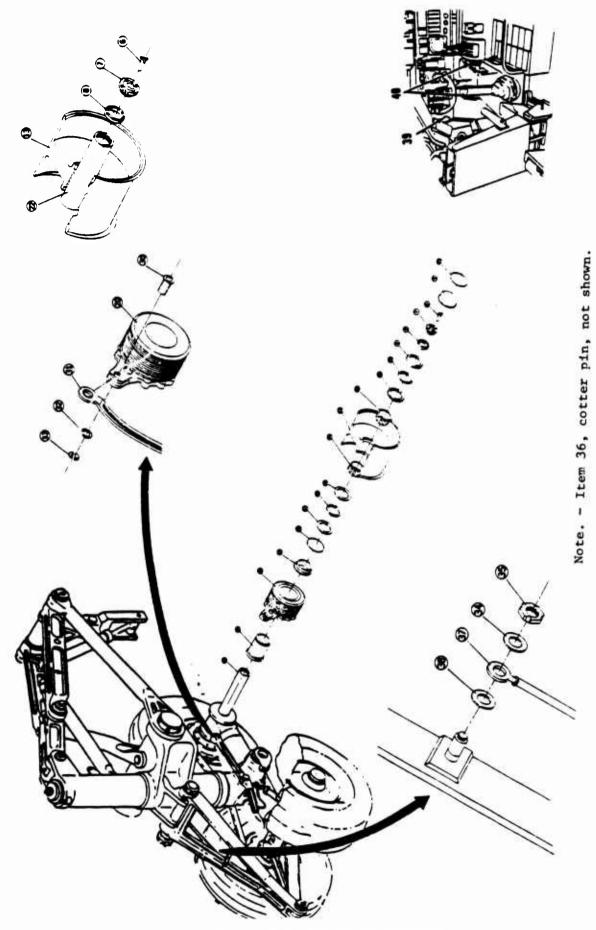


Figure 12 C-141 WHEEL AND BRAKE ASSEMBLY 150